

15th GERMAN LS-DYNA FORUM | October 15-17, 2018 Bamberg

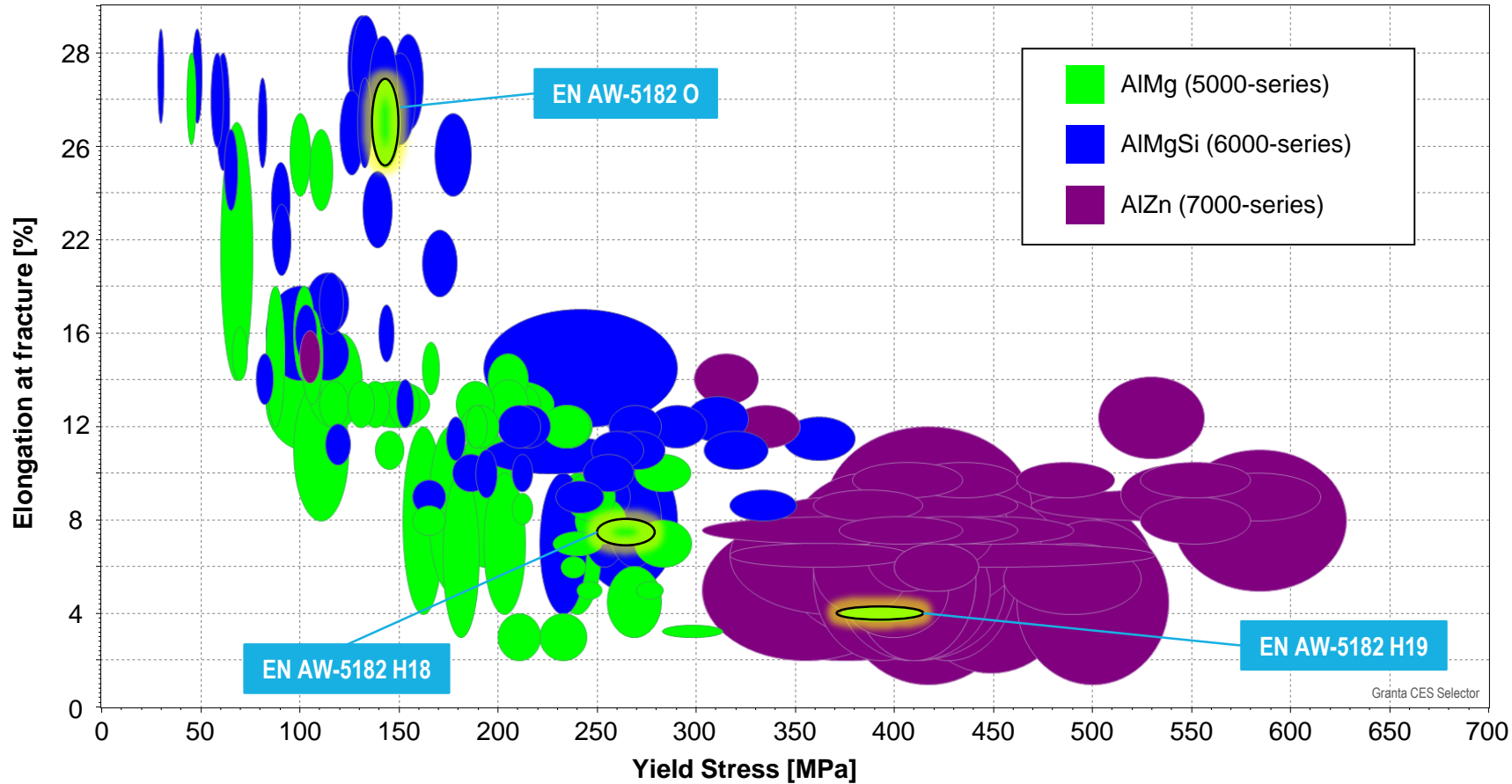
THE INFLUENCE OF DAMAGE

ACCUMULATION ON FAILURE PREDICTION

**A COMPARATIVE ASSESSMENT OF *MAT_224 AND *MAT_024 + GISSMO
FOR THE APPLICATION IN NON-ISOTHERMAL SHEET METAL FORMING**

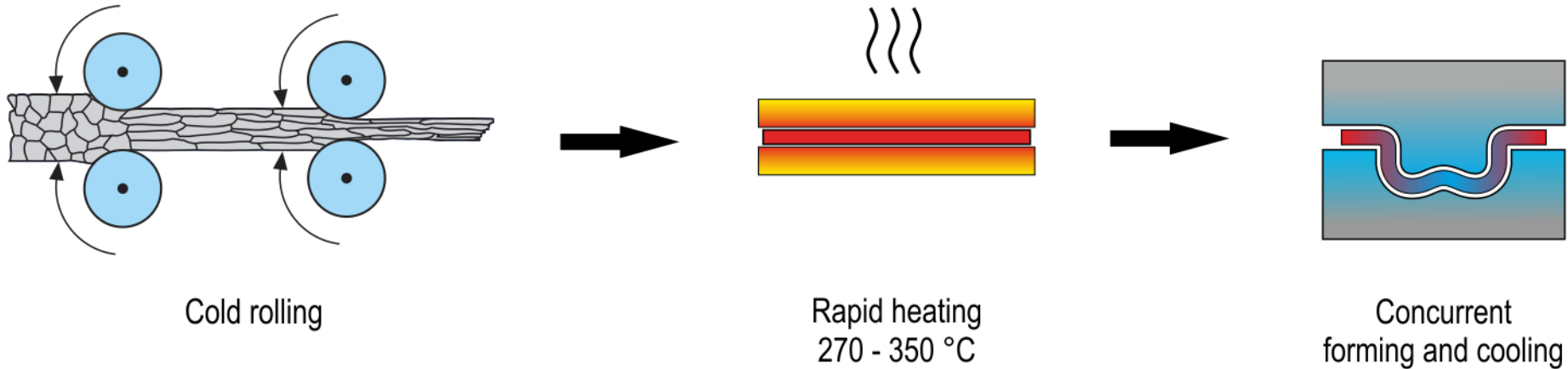
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Mechanical properties of typical automotive aluminum alloys.



- 5000-series aluminum alloys cover a wide range of mechanical properties
- Strengthening of 5000-series aluminum alloys is driven by solute hardening
- The formability of work-hardened H18 and H19 sheet materials is highly limited

FLASH FORMING PROCESS (FFP).



Motivation:

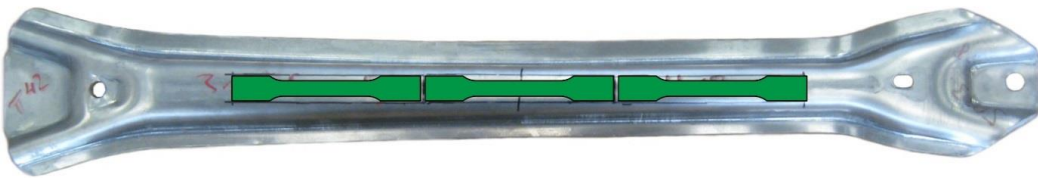
- Enabling the formability of work hardened AlMg sheet without severe recovery or recrystallization
- Substitution of precipitation hardening AlMgSi alloys by severe pre-strained AlMg sheet metals
- Process chain shortening, cost- and time-efficient component manufacturing

FLASH FORMING PROCESS: Forming trial of a door beam.

Wrought material: EN AW-5182 H18 t = 3.50 mm

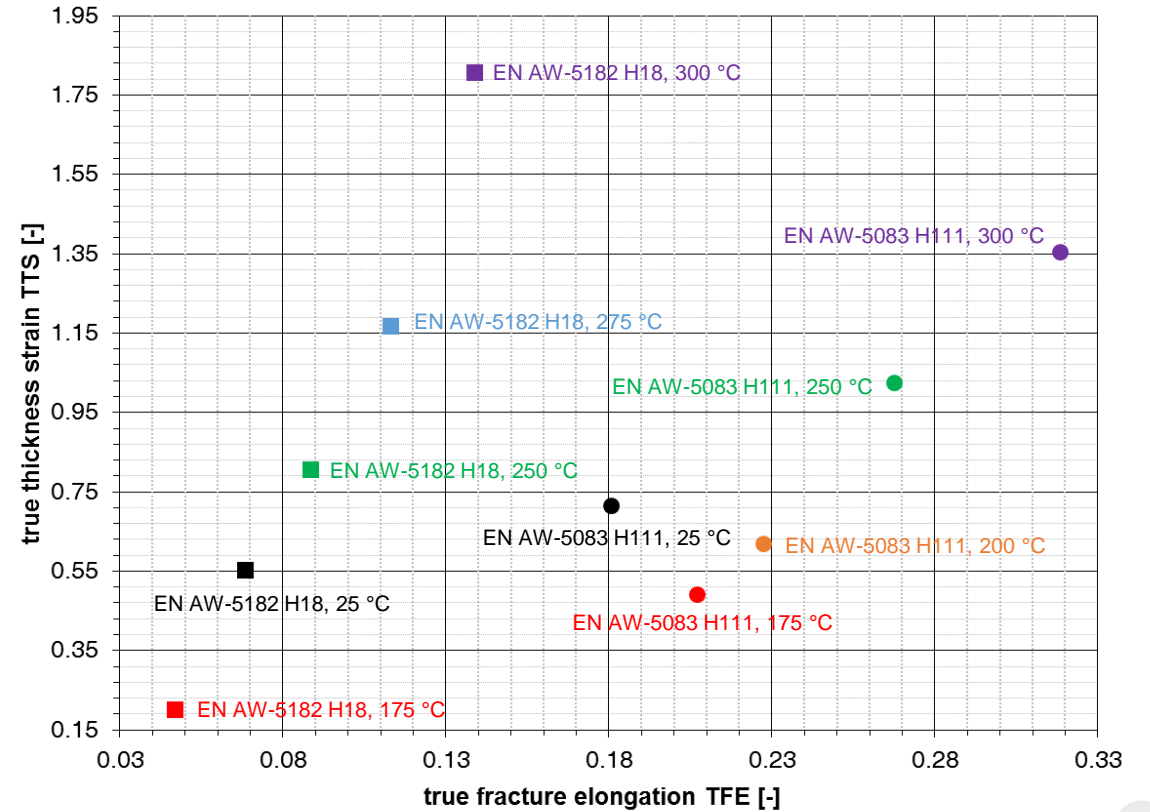
T1: RT

T2 > 300 °C



Parameter set	Heat treatment prior forming	Mean material properties after forming		
		YS [MPa]	UTS [MPa]	A ₃₀ [%]
T1	RT	349	398	9.0
T2	> 300 °C	285	359	12.0

local ductility vs. temperature



Camberg et al. (2018 a), Camberg et al. (2018 b)

Ductile failure prediction at non-isothermal conditions.

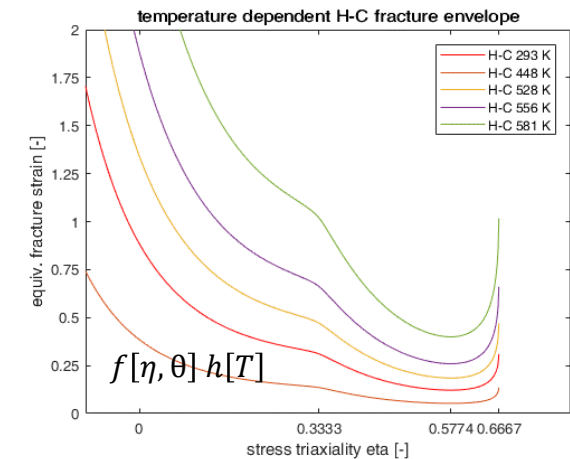
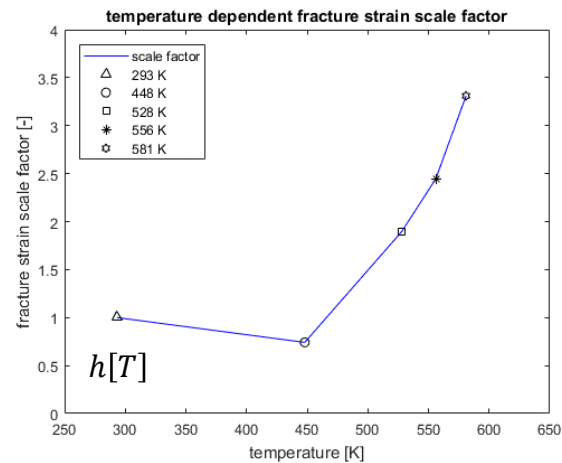
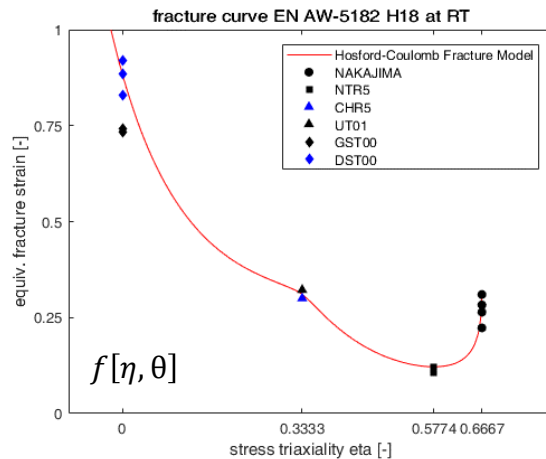
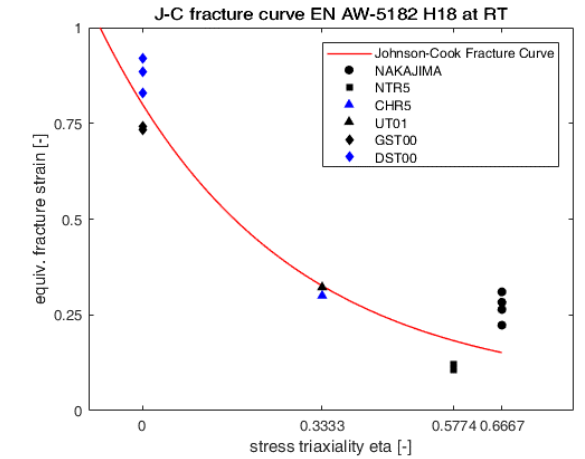
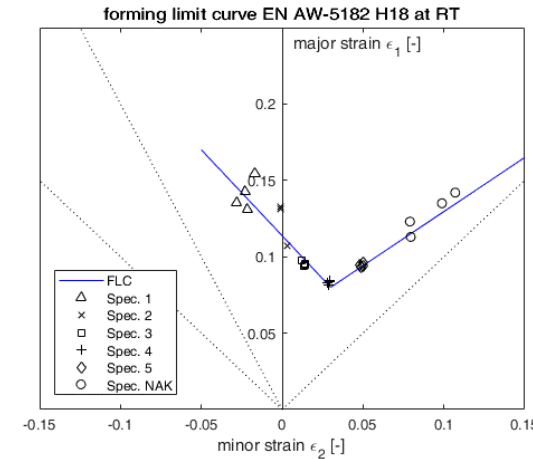
Forming Limit Curve: not suitable for shear fracture and tight radii bending, post-processing based

Johnson-Cook (1983) fracture initiation model:

$$\bar{\epsilon}_f^p[\eta, \dot{\epsilon}_p, T] = (D_1 + D_2 \exp(D_3 \eta))(1 + D_4 \ln(\dot{\epsilon}_p))(1 + D_5 \bar{T})$$

Buyuk (2013) – a general tabulated form of the J-C model (implemented as *MAT_224):

$$\bar{\epsilon}_f^p = f[\eta, \theta] g[\dot{\epsilon}_p] h[T] i[l_{el}]$$



Comparative assessment: *MAT_224 vs. *MAT_024 + GISSMO.

***MAT_224**: a general tabulated form of the J-C model based on *Buyuk (2013)*:

- J_2 based plasticity with strain rate effects and thermal softening:

$$f[\sigma, \bar{\epsilon}_p, T] = \bar{\sigma}_{vM} - k[\bar{\epsilon}_p, \dot{\bar{\epsilon}}_p, T] = \sqrt{\frac{3}{2} \sigma' : \sigma' - k_1[\bar{\epsilon}_p, \dot{\bar{\epsilon}}_p] k_t[\bar{\epsilon}_p, T]}$$

- Stress-state dependent fracture criterion with strain rate and thermal effects:

$$\bar{\epsilon}_f^p = f[\eta, \theta] g[\dot{\bar{\epsilon}}_p] h[T] i[l_{el}]$$

- Linear damage accumulation:

$$D = \left(\frac{d\bar{\epsilon}_p}{\bar{\epsilon}_f^p[\eta, \theta, \dot{\bar{\epsilon}}_p, T, l_{el}]} \right), \quad D = 1 \Rightarrow \text{Fracture}$$

- Triaxiality definition (!):

$$\eta = \frac{-I_1}{3\sqrt{3}J_2}$$

- Temperature increase due to plastic work:

$$\Delta T(\epsilon_p) = \frac{\beta}{C_p \rho} \int_0^{\epsilon_p} \sigma d\epsilon_p$$

***MAT_024 + GISSMO (*MAT_ADD_EROSION, IDAM = 1)**:

- J_2 based plasticity with strain rate effects:

$$f[\sigma, \bar{\epsilon}_p] = \bar{\sigma}_{vM} - k[\bar{\epsilon}_p, \dot{\bar{\epsilon}}_p] = \sqrt{\frac{3}{2} \sigma' : \sigma' - k[\bar{\epsilon}_p, \dot{\bar{\epsilon}}_p]}$$

- Damage accumulation, arbitrarily selectable damage exponent:

$$D = \left(\frac{d\bar{\epsilon}_p}{\bar{\epsilon}_f^p[\eta, \theta, \dot{\bar{\epsilon}}_p, l_{el}]} \right)^n, \quad D = 1 \Rightarrow \text{Fracture}$$

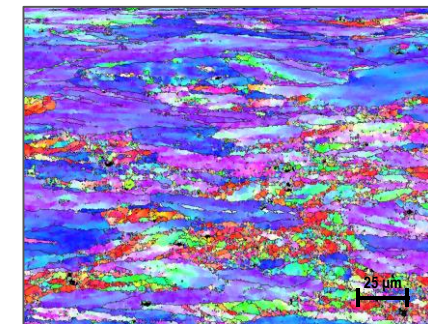
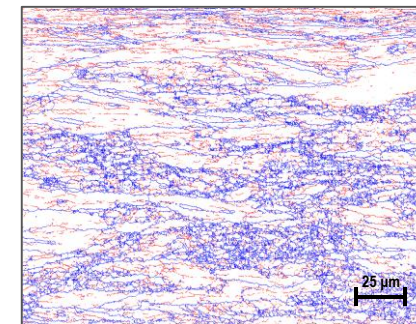
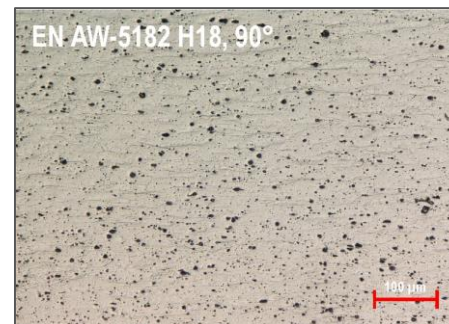
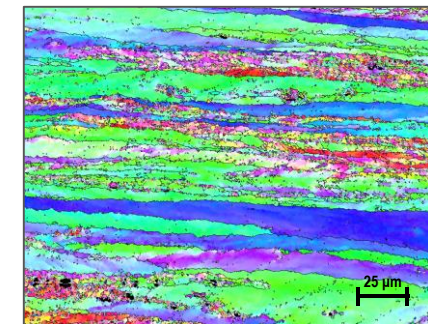
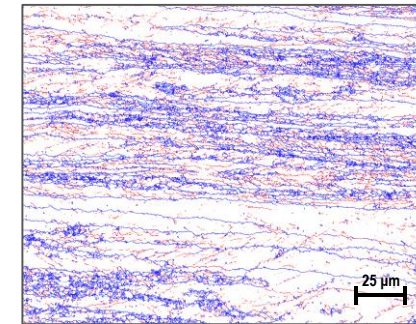
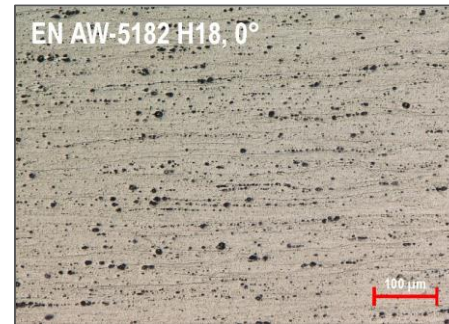
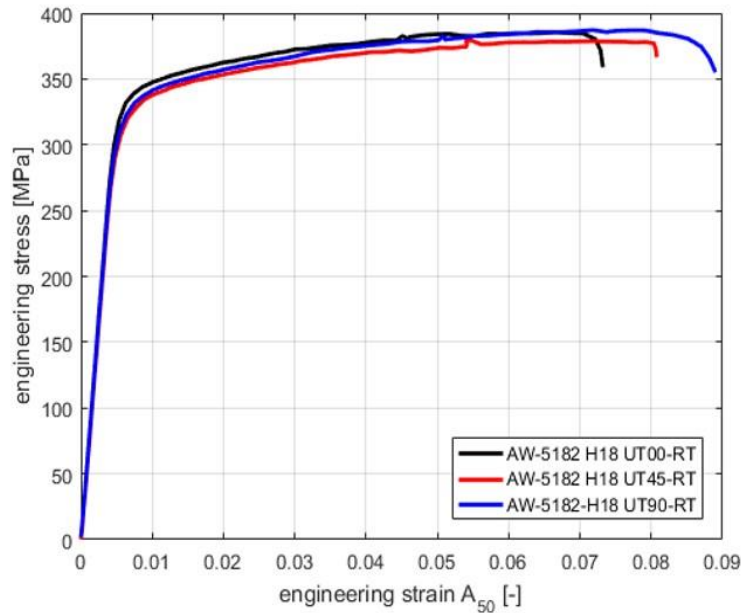
- Instability measure accumulation:

$$F = \left(\frac{d\bar{\epsilon}_p}{\bar{\epsilon}_{crit}^p[\eta, \theta]} \right)^n, \quad F = 1 \Rightarrow \text{Coupling of damage to stress tensor}$$

- Stress tensor degradation, arbitrarily selectable fading exponent:

$$\bar{\sigma}_{eff} = \bar{\sigma} \left(1 - \left(\frac{D - D_{crit}}{1 - D_{crit}} \right)^m \right), \quad D_{crit} = D(F = 1)$$

Material characteristics of EN AW-5182 H18 at room temperature.



RD [°]	E [MPa]	YS [MPa]	UTS [MPa]	A _g [-]	A [-]	R _{0.01-0.03} [-]	YS/YS ₀₀ [MPa]	R/R ₀₀ [-]
00	68.017	335.42	385.42	0.061	0.066	0.470	1	1
45	66.108	322.27	379.65	0.066	0.073	1.026	0.978	2.18
90	67.136	326.59	387.16	0.066	0.080	1.091	0.982	2.32

Camberg et al. (2018 b)

Hardening curve: Inverse fit of experimental data to account for instability.

1. Fitting experiment data in the pre necking region by a strictly monotonically increasing function:

$$\sigma_{pre\ neck.} = \sigma_y + a * (1 - e^{-b*\epsilon^p}) + c * (1 - e^{-d*\epsilon^p}) + e * (1 - e^{-f*\epsilon^p}) + g * (1 - e^{-h*\epsilon^p})$$

2. Hardening curve extrapolation in post necking region by Power Law / Swift Hardening Law

$$\sigma_{post\ neck.} = k (\epsilon^p + \epsilon_{seam})^n$$

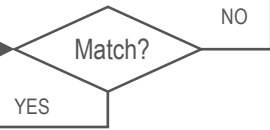
$$k = \sigma_{neck.} * \left(\frac{\sigma_{neck.} * n}{\sigma'_{pre\ neck.}(\epsilon_{neck.})} \right)^{-n}$$

$$\epsilon_{seam} = \frac{\sigma_{neck.} * n}{\sigma'_{pre\ neck.}(\epsilon_{neck.})} - \epsilon_{neck.}$$

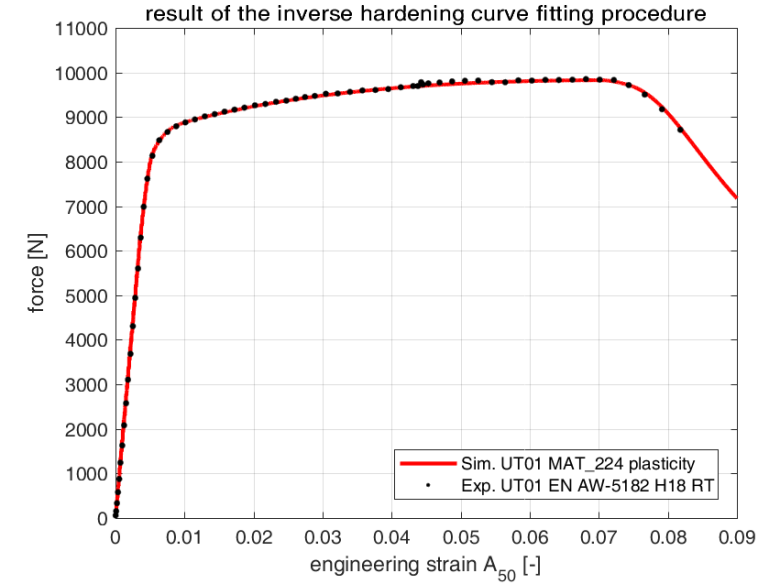
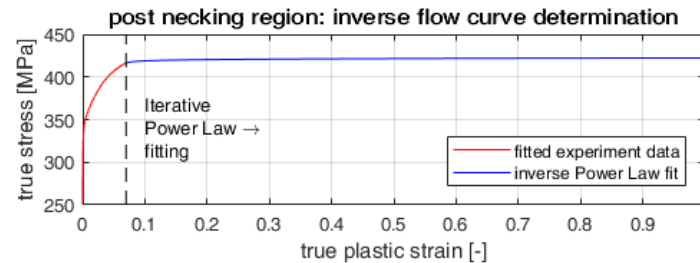
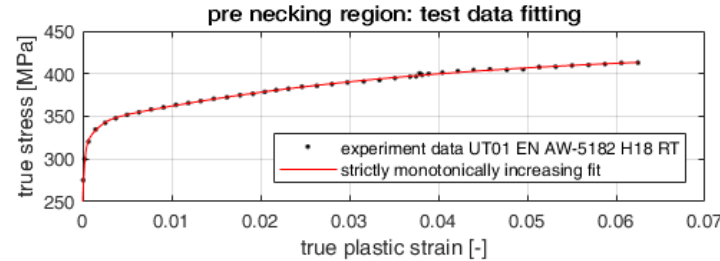
Adjustment of n

3. Finite Element Analysis

4. Validation with test data

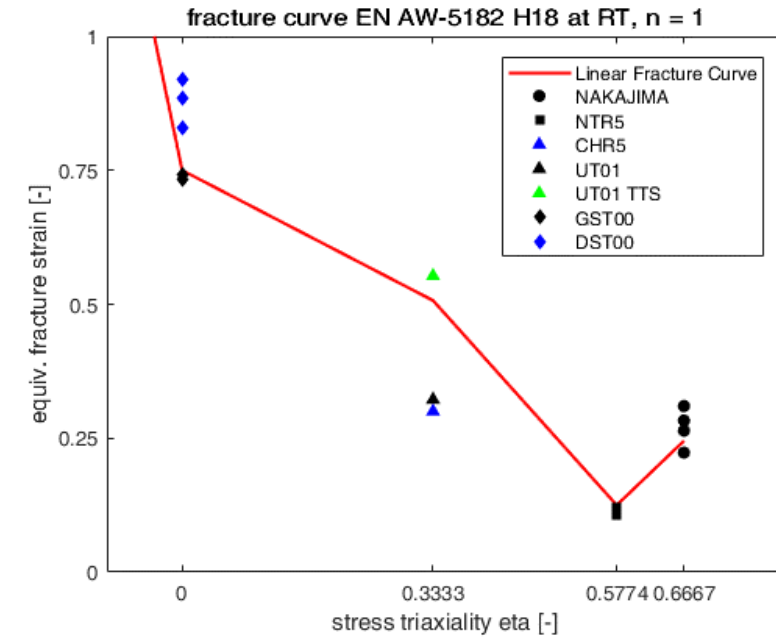
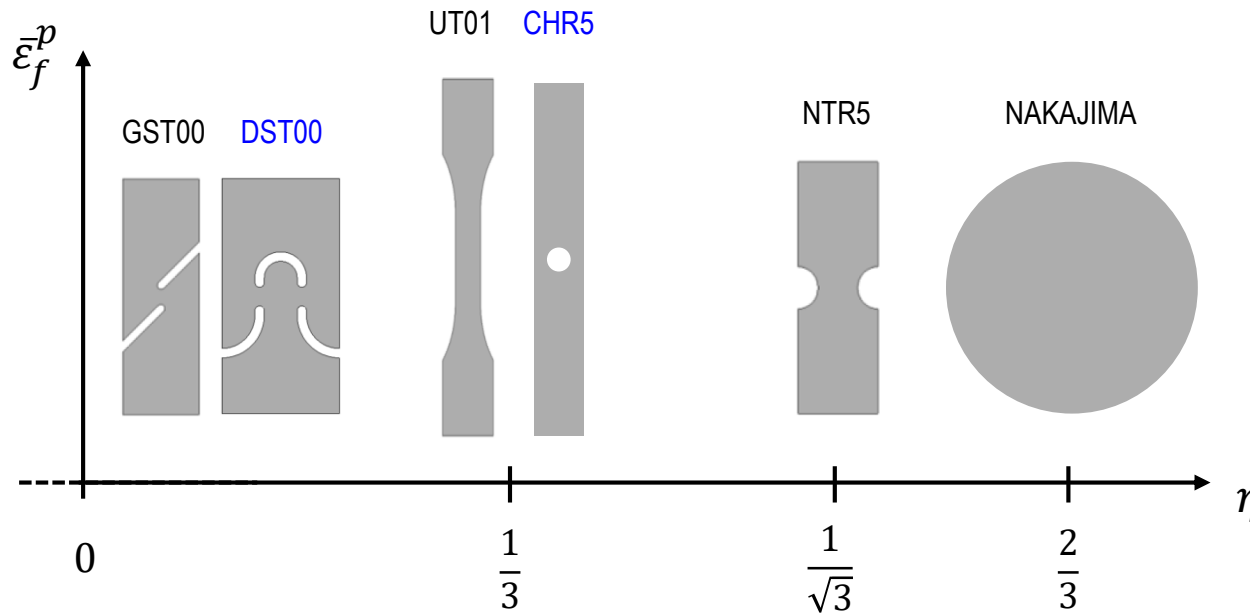


5. Done.



- To correctly predict fracture the stress-strain curve after necking has to match test data as good as possible
- Since *MAT_224 does not provide a coupling function between damage and plasticity (as available in GISSMO) the curve is fitted solely by the hardening potential after necking
- The adjustment is carried out by an inverse parameter identification by varying the parameter n of the hardening law

Fracture curve calibration: Inverse fit of experimental curves.



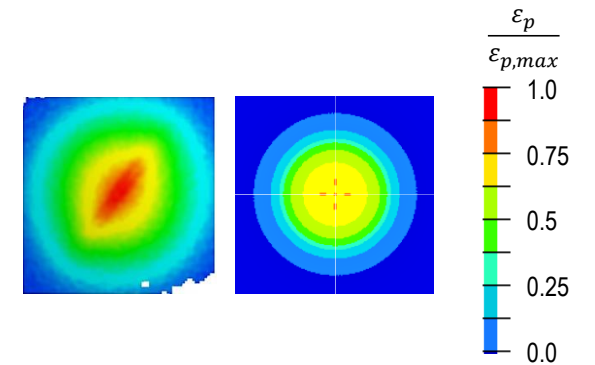
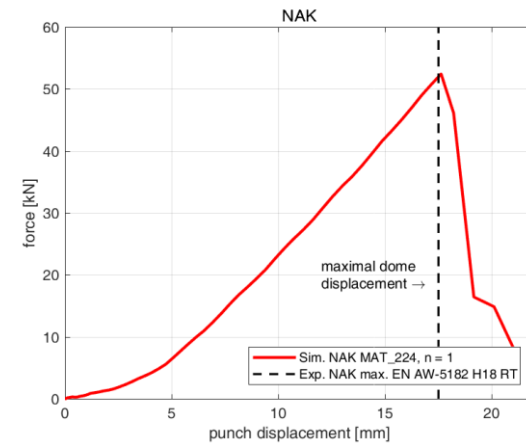
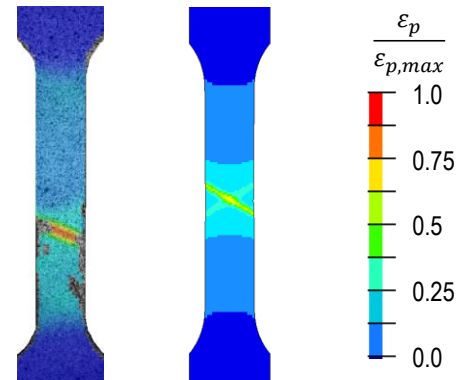
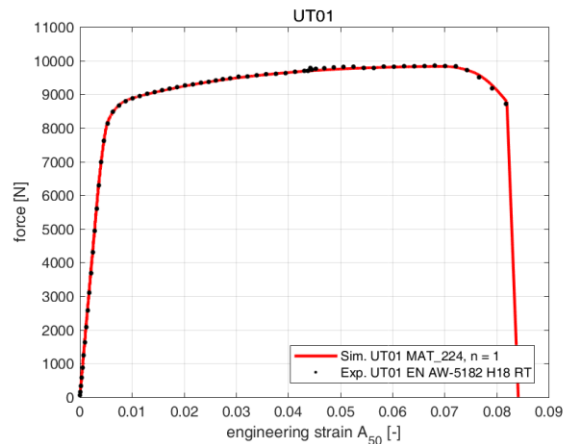
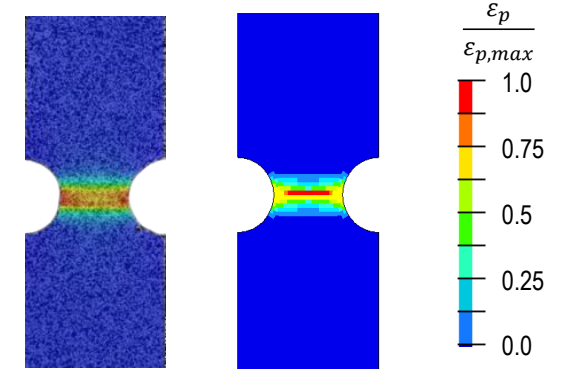
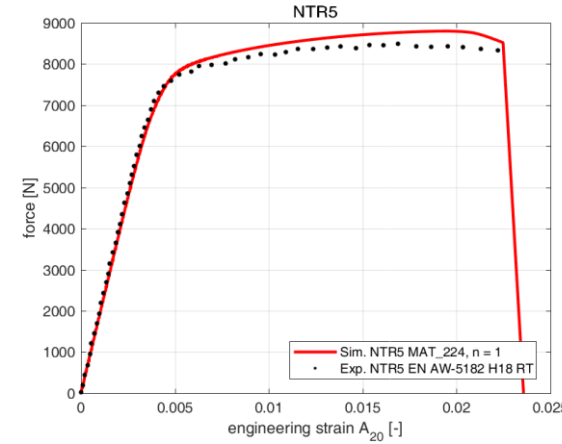
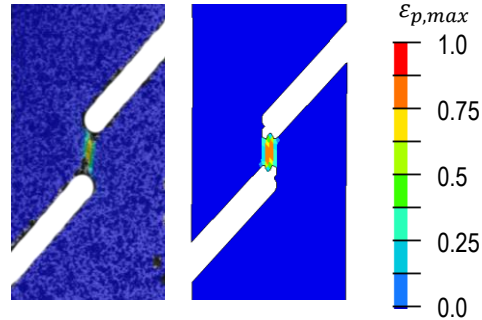
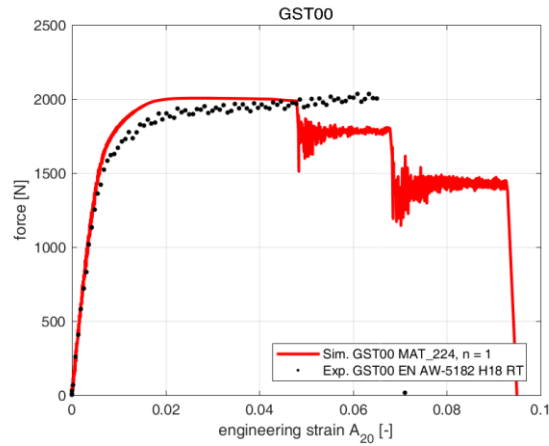
R7.1.3, ELFORM = 16, NIP = 7, l_{el} = 0.5 mm

True Thickness Strain (TTS):

$$TTS = \ln\left(\frac{t_f}{t_o}\right)$$

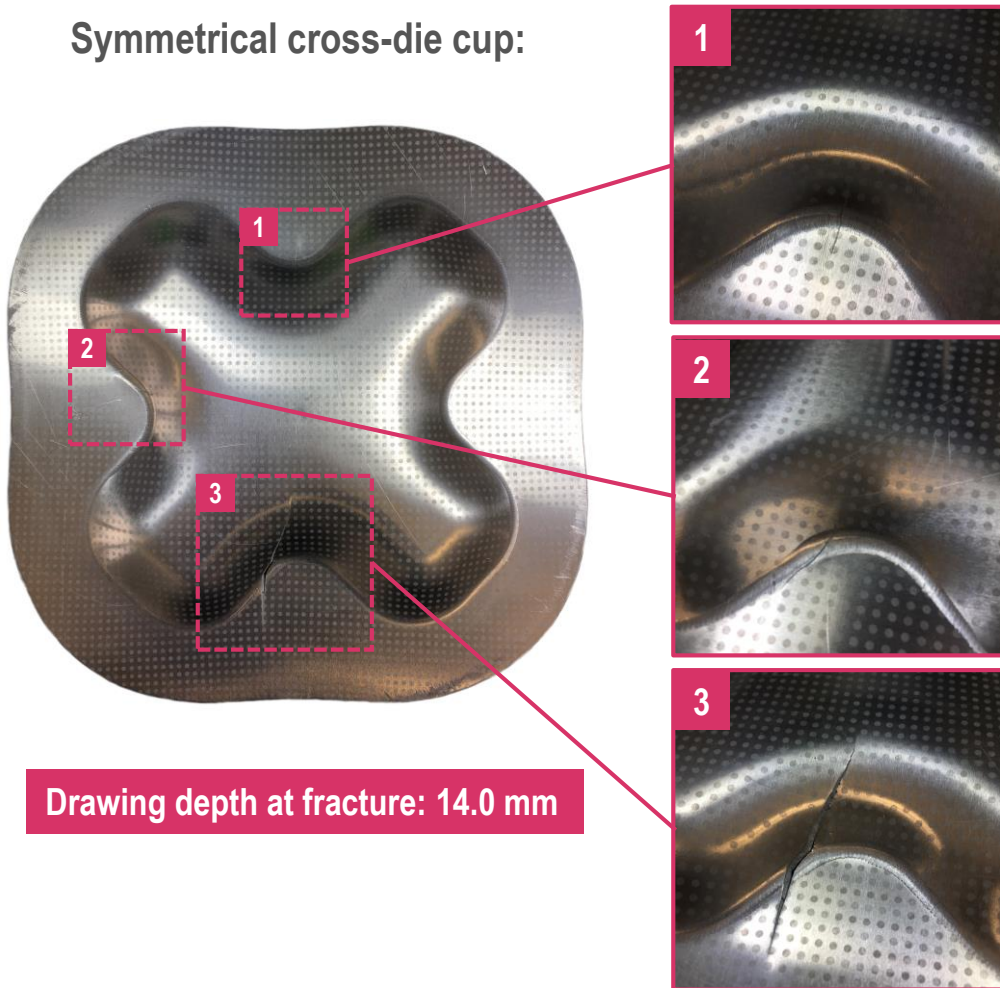
- As proposed by *Andrade et al. (2014)* and *Andrade et al. (2016)* the basic fracture curve (RT) is determined by an inverse fit of experimental force-displacement curves.
- In analogy to *Andrade et al. (2014)* and *Heibel et al. (2017)* the fracture envelope is based on a linear interpolation between the characteristic triaxialities.
- The True Thickness Strain (TTS) was additionally invoked to validate the fitting point for UT01

***MAT_224 fracture curve calibration: Inverse fit of experimental force-displacement curves.**

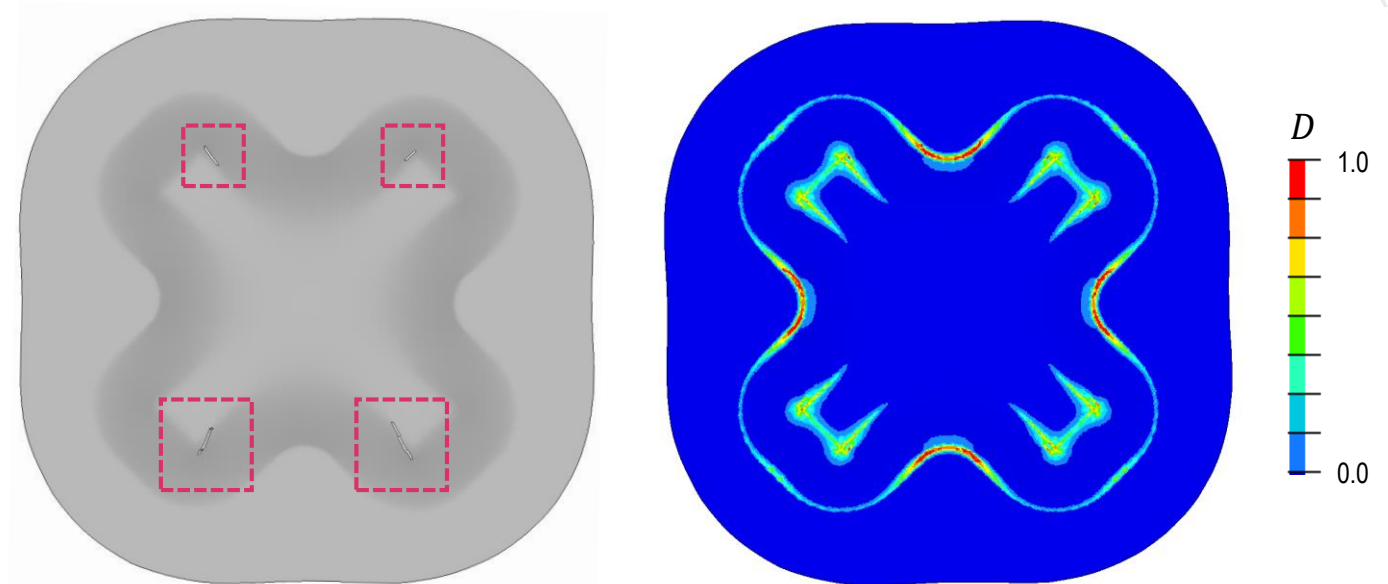


***MAT_224: Validation on a cross-die cup at room temperature.**

Symmetrical cross-die cup:



***MAT_224 at room temperature (baseline):**



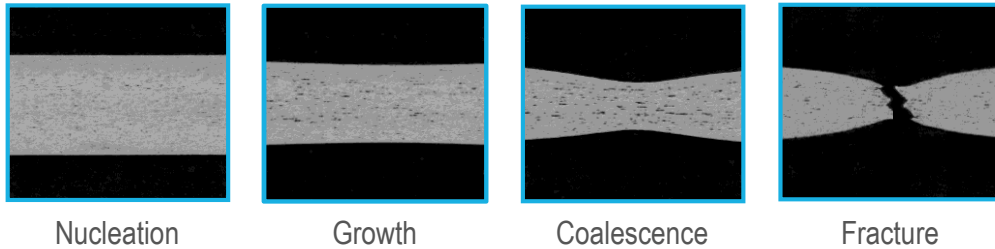
Drawing depth at fracture: 8.5 mm

Critical triaxialities: $\eta = \frac{1}{\sqrt{3}}$, $\eta = \frac{2}{3}$

- The calibrated *MAT_224 is not able to predict the critical drawing depth and fracture location with sufficient accuracy

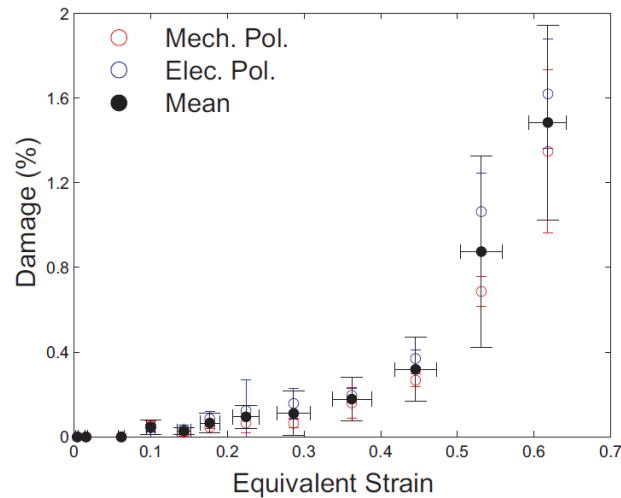
Damage accumulation: The influence of damage exponent on failure prediction.

Damage evolution of EN AW-5083 H111 at 300 °C:



- Experimental investigations, as e.g. presented by *Tasan (2010)*, show a non-linear evolution of damage
- *MAT_224 adopts a linear damage accumulation from the classical J-C model what could be a reason for underestimating the critical draw depth
- By setting `ECRIT = 0`, `DCRIT = 1.0e+06` and `FADEXP = 1.0e+06` an isothermal *MAT_224 with an arbitrary damage exponent is restored by *MAT_024 + GISSMO

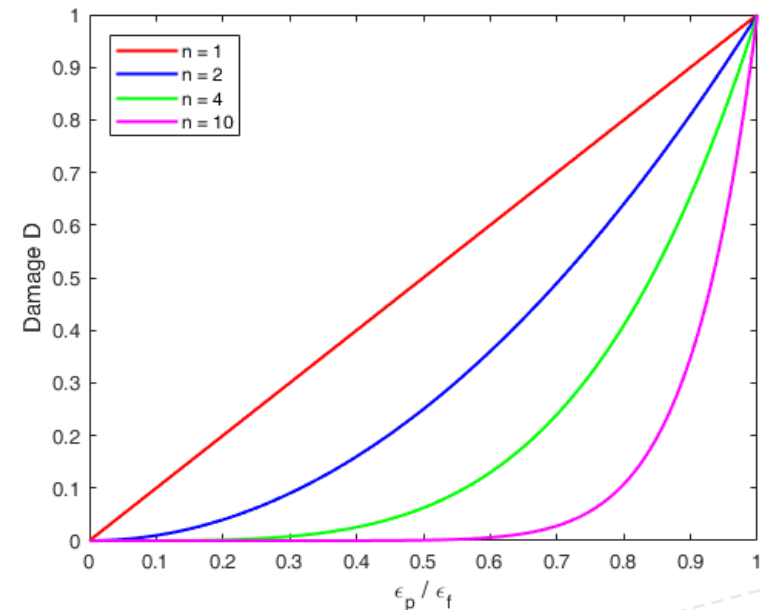
Damage evolution of a DP steel



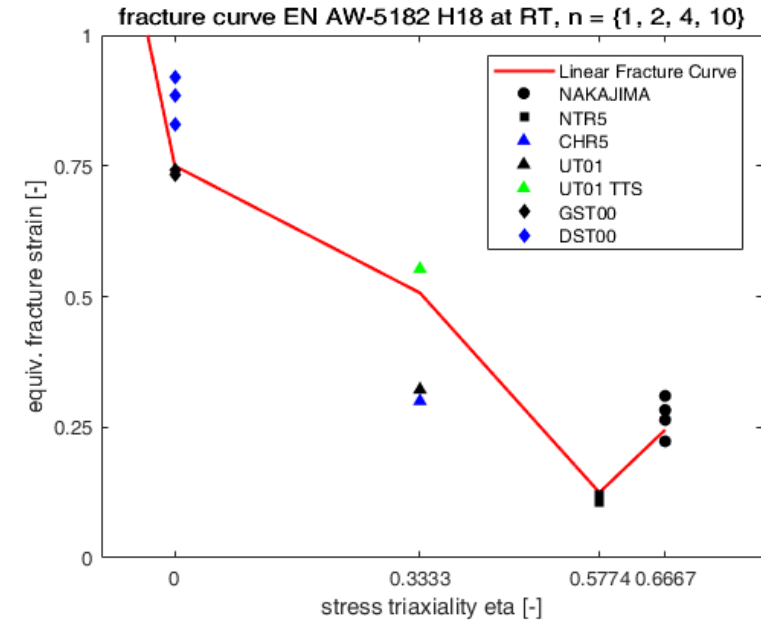
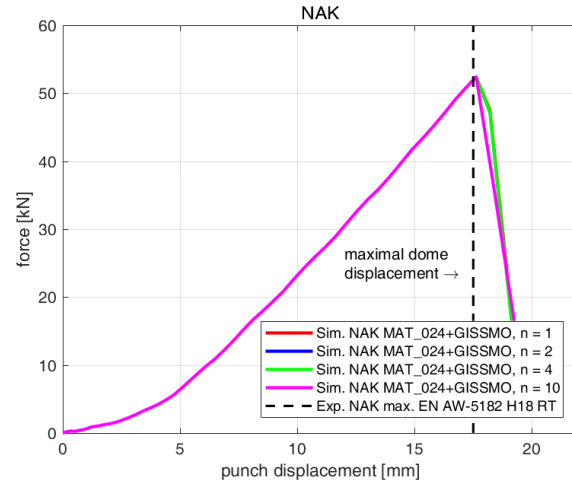
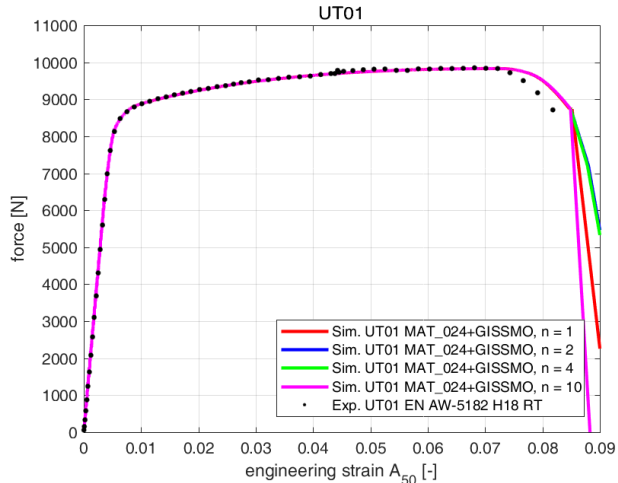
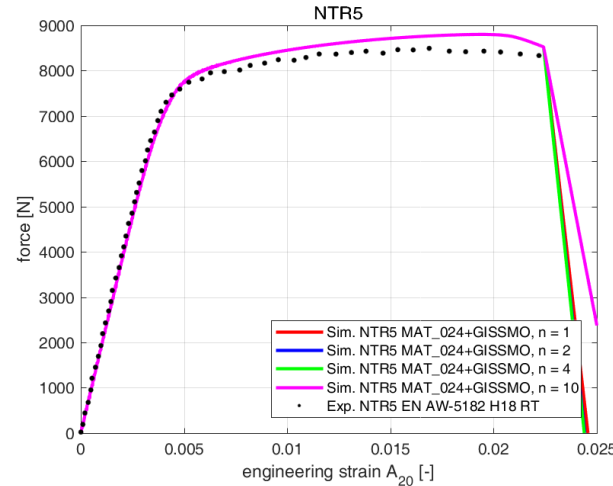
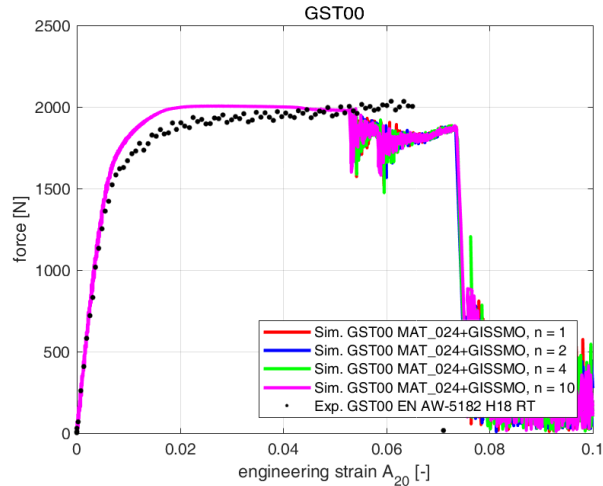
Tasan CC (2010) *Micro-mechanical characterization of ductile damage in sheet metal*
Dissertation Technische Universiteit Eindhoven

Damage accumulation:

$$D = \left(\frac{d\bar{\epsilon}_p}{\bar{\epsilon}_f^p[\eta]} \right)^n$$



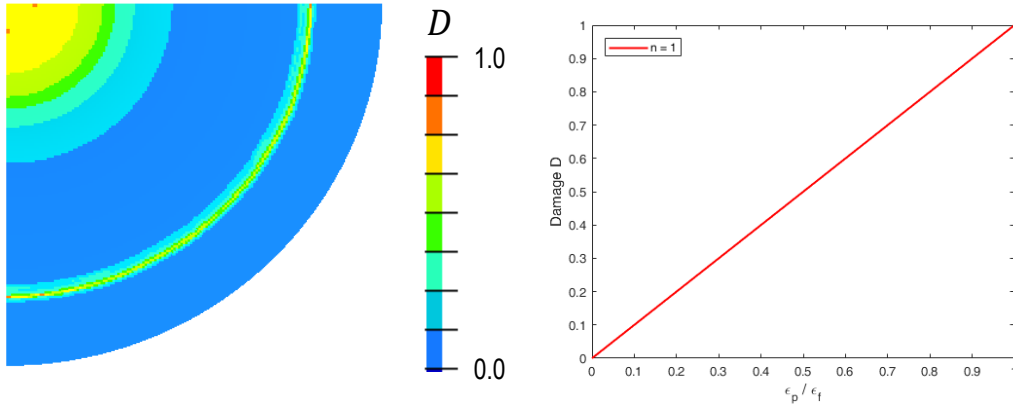
***MAT_024 + GISSMO fracture curve calibration for damage exponents $n = \{1, 2, 4, 10\}$.**



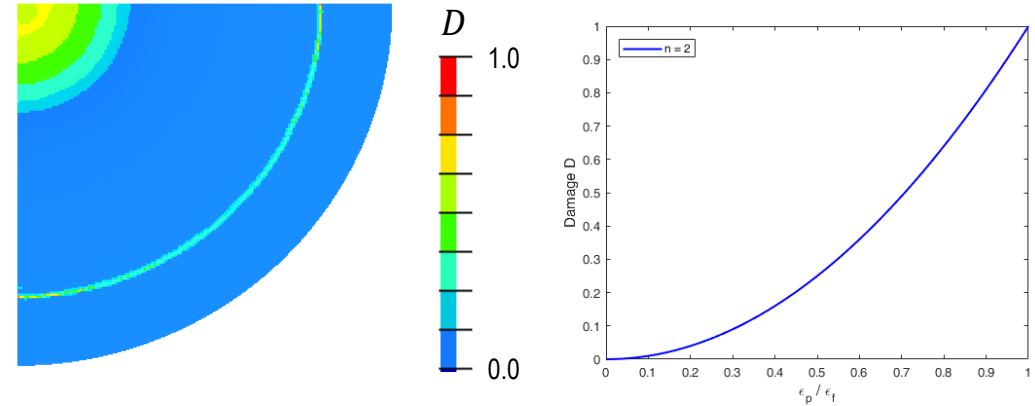
- The fracture curve remains the same as for *MAT_224, no additional calibration is required
- The plasticity of *MAT_224 and *MAT_024 show a slight mismatch for UT01
- With *MAT_024 + GISSMO the shear specimen (GST00) matches the test data better
- The damage exponent has no influence on the predicted drawing depth due to approximately linear strain paths of the calibrations specimens (for this material)

***MAT_024 + GISSMO $n = \{1, 2, 4, 10\}$: Damage evolution at equi-biaxial loading (NAK).**

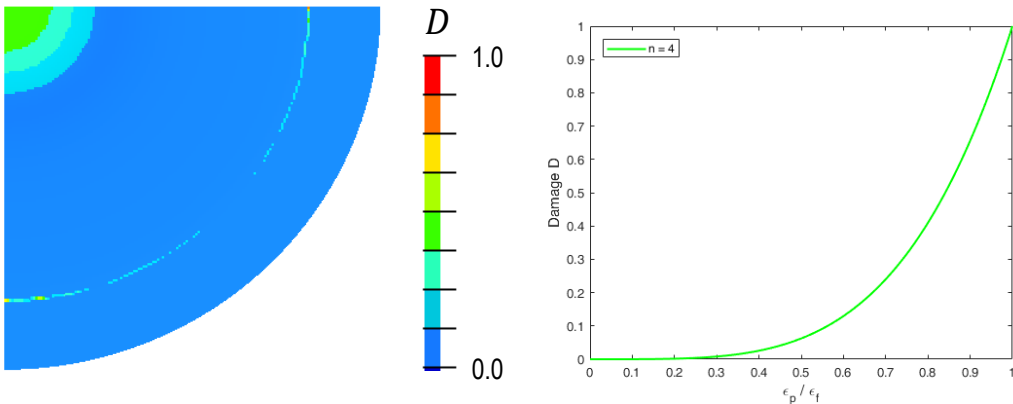
GISSMO, Damage exponent $n = 1$, punch displacement 17.6 mm



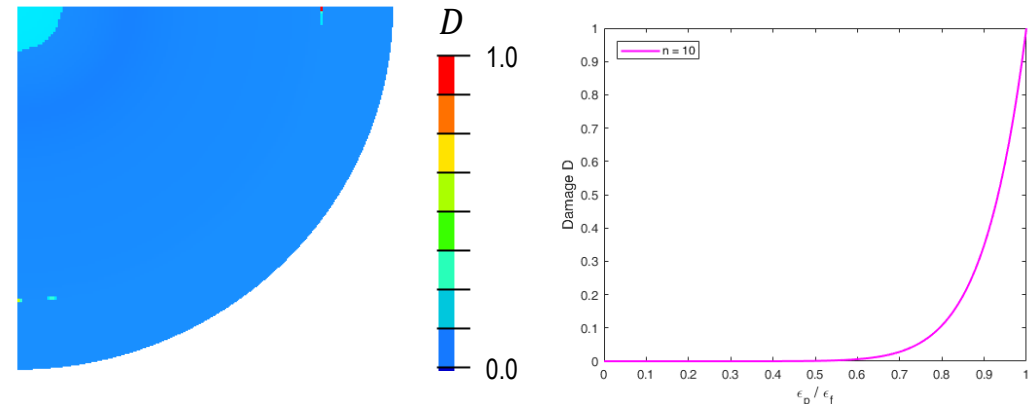
GISSMO, Damage exponent $n = 2$, punch displacement 17.6 mm



GISSMO, Damage exponent $n = 4$, punch displacement 17.6 mm

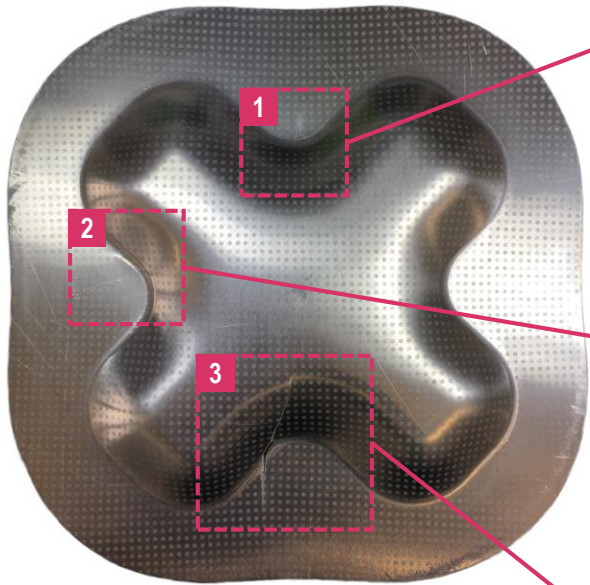


GISSMO, Damage exponent $n = 10$, punch displacement 17.6 mm

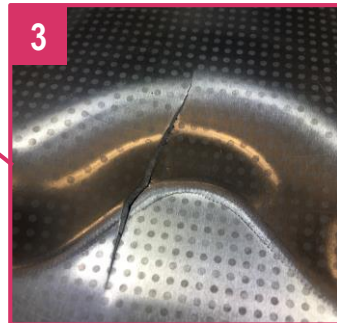
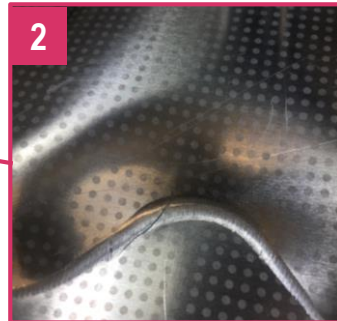


***MAT_024 + GISSMO $n = \{1, 2, 4, 10\}$: Validation on a cross-die cup at room temperature.**

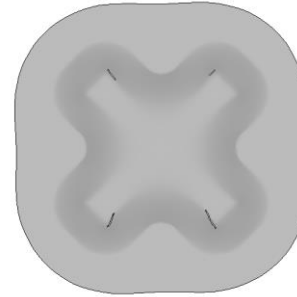
Symmetrical cross-die cup:



Drawing depth at fracture: 14.0 mm

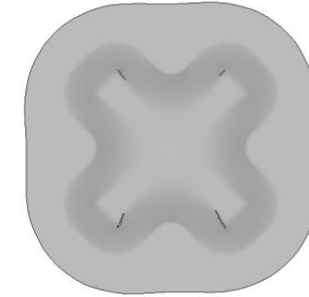


*MAT_024 + GISSMO, $n = 1$:



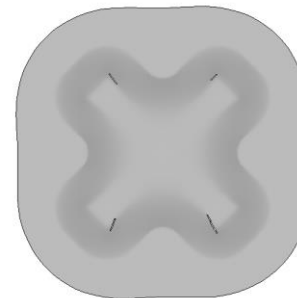
Drawing depth at fracture: 8.5 mm

*MAT_024 + GISSMO, $n = 2$:



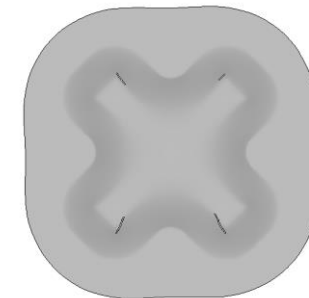
Drawing depth at fracture: 8.5 mm

*MAT_024 + GISSMO, $n = 4$:



Drawing depth at fracture: 8.5 mm

*MAT_024 + GISSMO, $n = 10$:

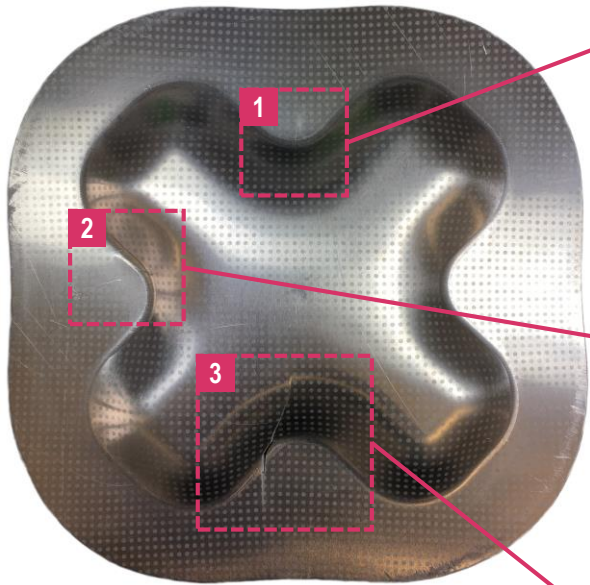


Drawing depth at fracture: 8.5 mm

► The damage exponent has no influence on the predicted drawing depth due to approximately linear strain paths

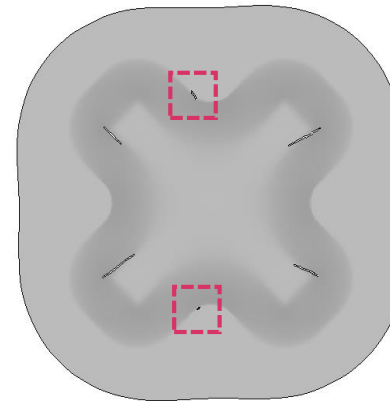
***MAT_024 + GISSMO: Invoking instability measure (F) and stress tensor degradation.**

Symmetrical cross-die cup:

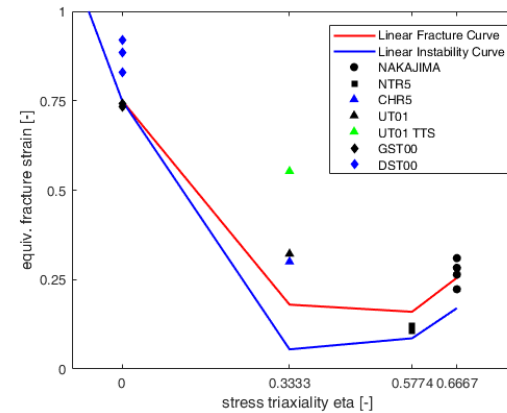


Drawing depth at fracture: 14.0 mm

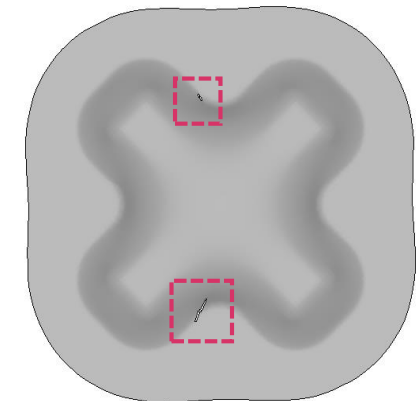
***MAT_024 + GISSMO (damage and instability), n = 2:**



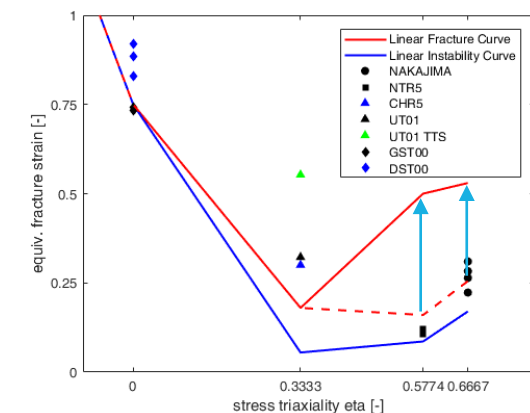
Drawing depth at fracture: 8.4 mm



***MAT_024 + GISSMO (damage and instability), n = 2, reversely fitted on cross-die:**



Drawing depth at fracture: 14.0 mm



Conclusions & Outlook.

- Novel processing routes enable the stamping of high-strength AlMg(Mn) automotive parts.
- The calibrated *MAT_224 is not able to predict the critical drawing depth and fracture location with sufficient accuracy.
- The tested material fails without pronounced necking, so the critical strain paths of the calibration specimens can be considered as approximately linear.
- For this reason, the damage exponent has no influence on the predicted onset of fracture which applies to both, the calibration samples and the cross-die cup.
- A precise modelling of the anisotropic yield characteristics of the investigated highly pre-strained material appears to be prerequisite for a correct fracture prediction. However, the experimental determination of EN AW-5182 H18 material parameters with conventional methods (e.g. Bulge test) is challenging.
- In future work the abilities of Barlat2000 (*MAT_133) and more advanced anisotropic yield loci models (e.g. non-associated plasticity) will be investigated in combination with GISSMO for their ability to predict fracture correctly at isothermal conditions. A model enrichment for non-isothermal conditions in analogy to *MAT_224 will follow after satisfactory results.
- However, since *MAT_224 is also available for shell elements, an instability behavior and coupling to the stress tensor, as implemented GISSMO, should be considered in the future. Alternatively GISSMO could be extended by a temperature dependent term.
- Furthermore, it should be questioned whether shell elements with 0.5 element edge length are reliable to simulate materials with $t > 2.0$ mm
- The failure prediction of highly work hardened AlMg alloys at isothermal (RT) conditions is already challenging what makes the non-isothermal failure prediction even more thrilling!

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