

Infotag

“Nichtlineare Optimierung und stochastische Analysen”
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Multidisciplinary Optimization using the Successive Response Surface Method

Heiner Müllerschön, Nielen Stander
hm@dynamore.de, nielen@lstc.com

Topics

- Introduction
- LS-OPT: Application of the Successive Response Surface Method (SRSM)
- Example: Multidisciplinary Optimization (MDO)

Introduction

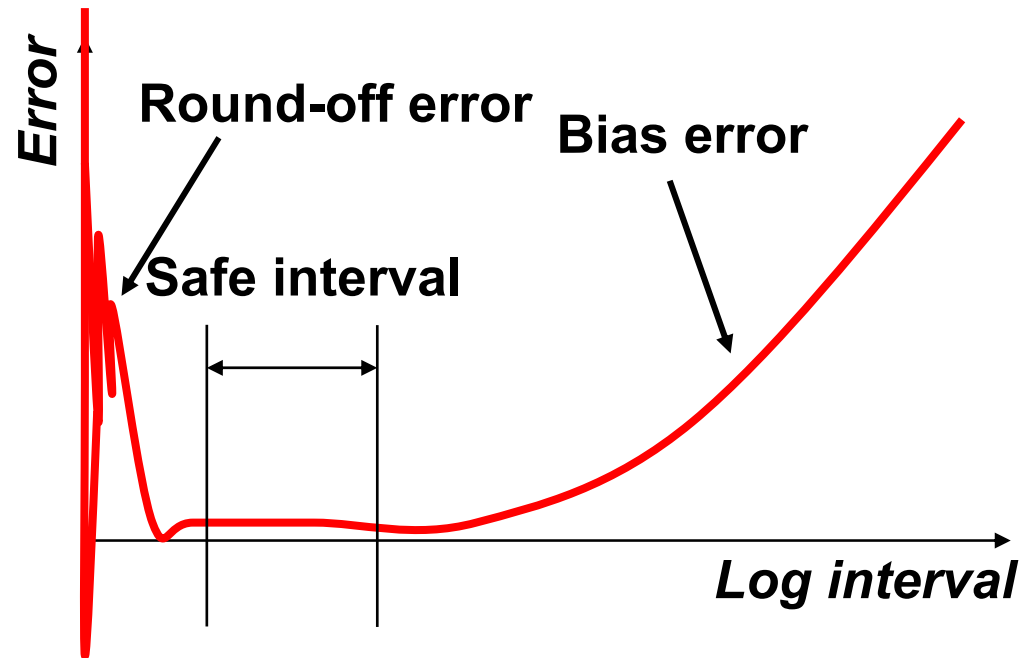
What is LS-OPT?

- **LS-OPT is an environment to explore automatically the design space and find an optimum design**
- **LS-OPT is a product of LSTC (Livermore Software Technology Corporation)**
- **LS-OPT is based on the Successive Response Surface Method (SRSM). Statistical approaches (Robustness Analysis) and genetic algorithms (Discrete Methods) will be implemented in near future**
- **LS-OPT provides a graphical user interface (GUI)**
- **LS-OPT can be linked to any simulation code, but it is perfect suitable in combination with LS-DYNA**

LS-OPT: Application of the SRSM

Why Response Surface Method and not Gradient Based Methods?

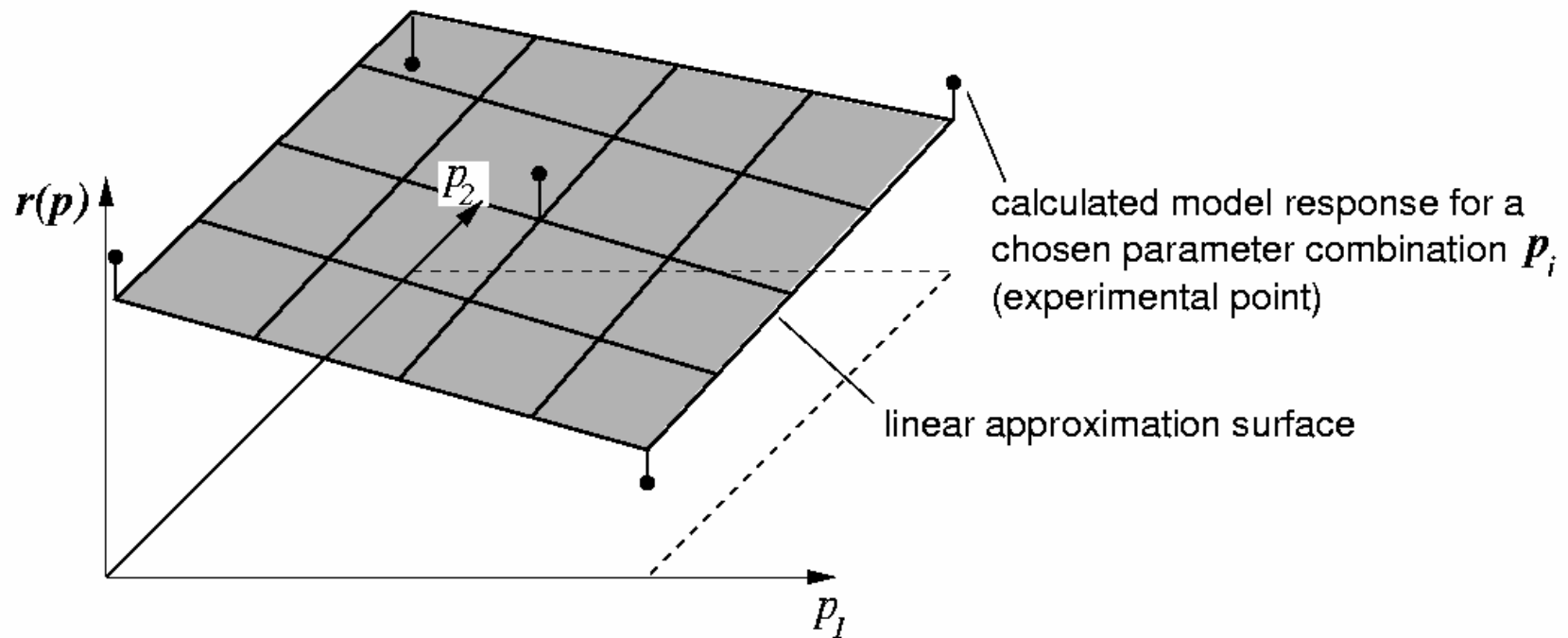
- **Highly Nonlinear Problems**
- **Local Sensitivities may lead to local optimums**
- **Difficulties by the Computation of Numerical Gradients**
 - **If the perturbation interval is too large: loose accuracy**
 - **If the perturbation interval is too small: find spurious gradients**



LS-OPT: Application of the SRSM

SRSM: How does it work?

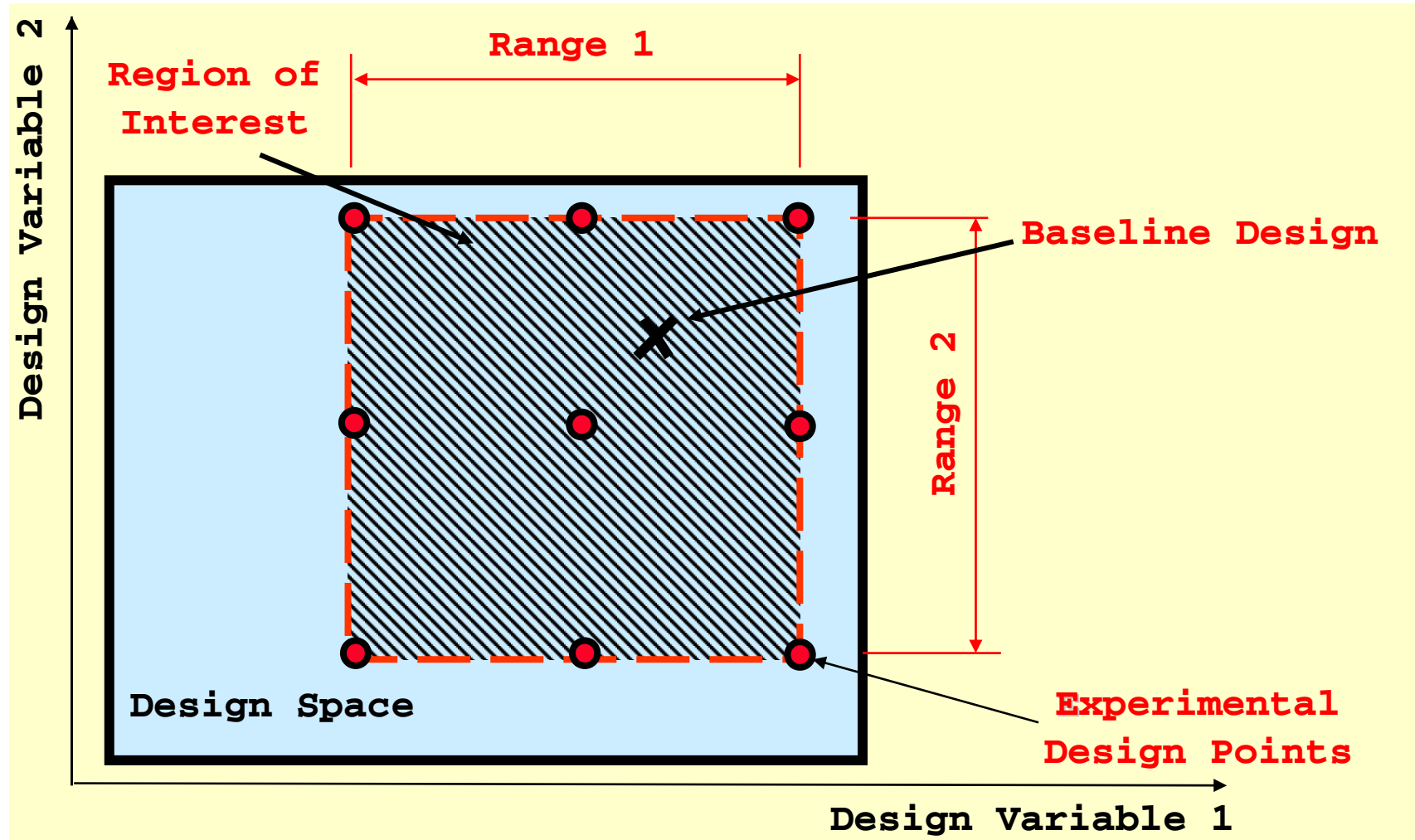
- **Design surfaces are fitted through points in the design space to form approximate optimization problem**



- **The idea is to find surfaces with the best predictive capability**

LS-OPT: Application of the SRSM

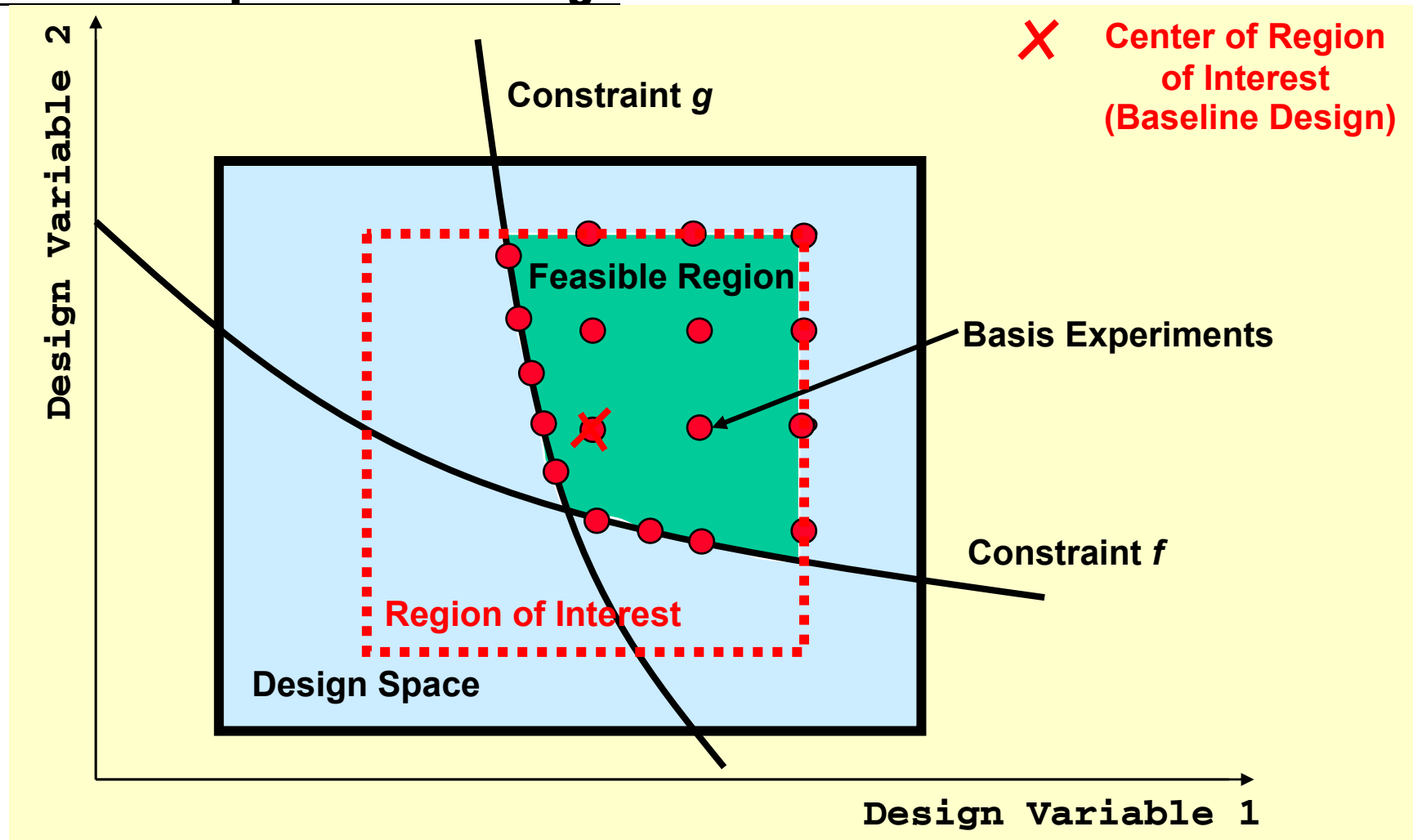
Design Space, Region of Interest & Experimental Design Points



Optimization using the Successive Response Surface Method

LS-OPT: Application of the SRSM

Feasible Experimental Design

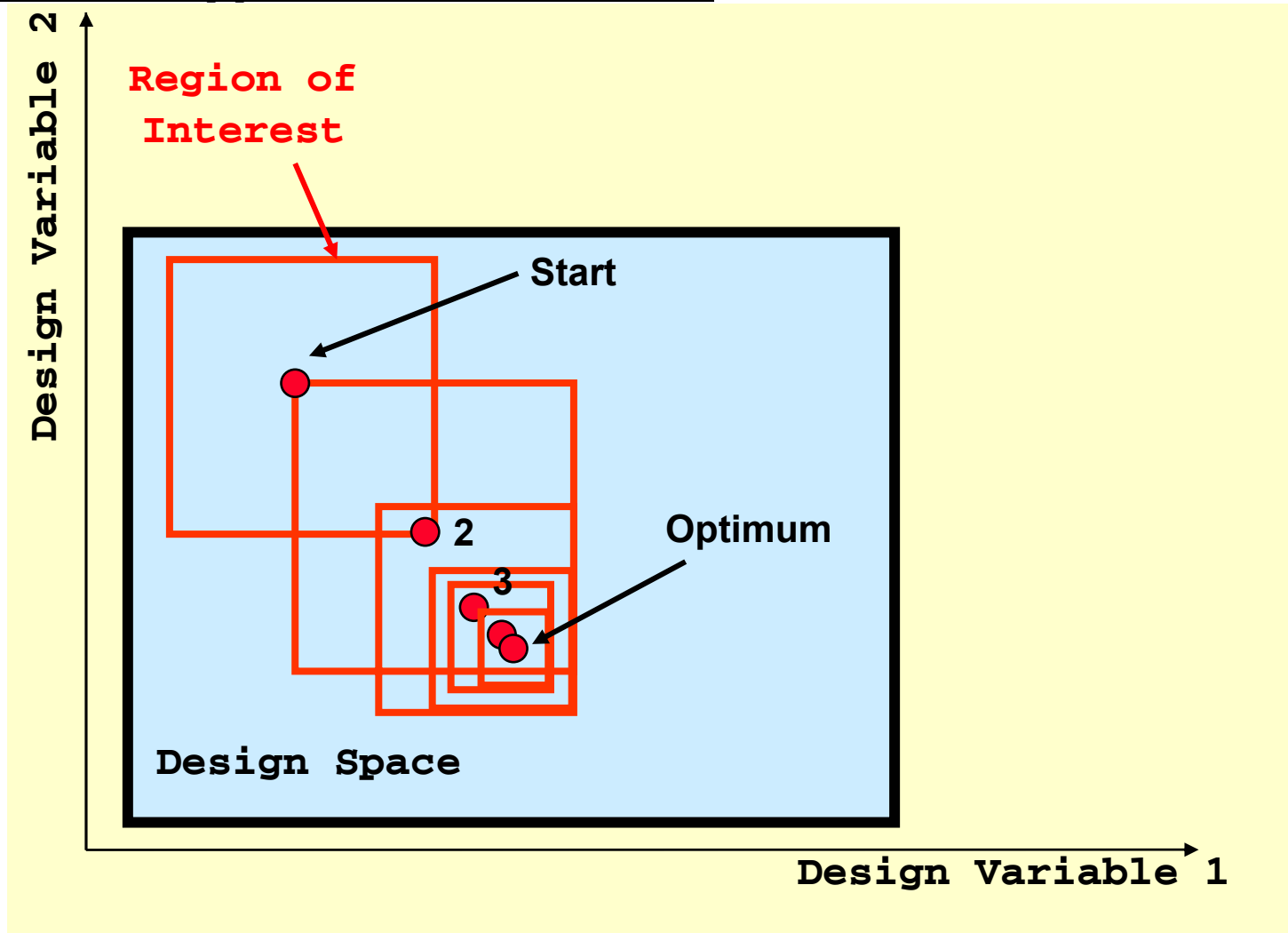


Optimization using the Successive Response Surface Method



LS-OPT: Application of the SRSM

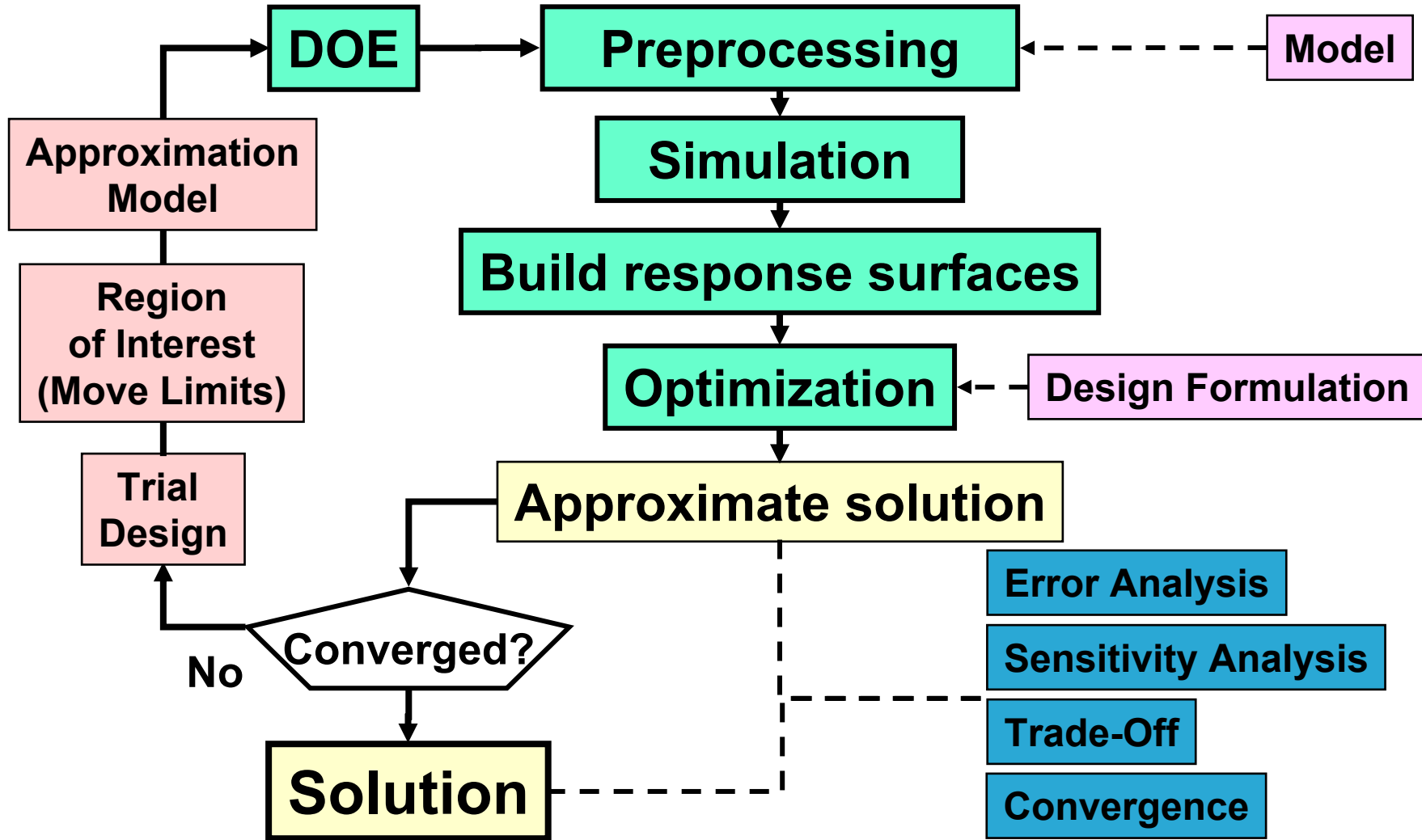
Successive Approximation Scheme



Optimization using the Successive Response Surface Method

LS-OPT: Application of the SRSM

The Optimization Prozess



Optimization using the Successive Response Surface Method

LS-OPT: Application of the SRSM

Graphical User Interface

The screenshot displays the LS-OPT software interface. At the top is a menu bar with options: File, Help, Info, Solver, ExpDesign, Histories, Responses, Constraints, Objective, Run, View. The main window is divided into two primary sections.

EXPERIMENTAL DESIGN

Response Surface Approximation: Linear (dropdown menu)

Point Selection Scheme: D-Optimal (dropdown menu)

Points per Variable for Basis: 4 points (dropdown menu)

Number of Experiments (one simulation per): 8 (input field) Compute Default

Leave blank to always use default.

Design Space Plot

The plot shows Design Variable 2 on the vertical axis and Design Variable 1 on the horizontal axis. A red dashed rectangle defines the **(Region of Interest) DESIGN SPACE**. The corners are labeled: Min 1 (left), Max 1 (right), Min 2 (bottom), and Max 2 (top). A red 'x' marks a point within the design space.

Design Variables Table

| Type | Name | Starting | Range | Minimum | Maximum | Transformation |
|-----------|--------|-------------------------------|-------|---------|---------|----------------|
| Variable | var130 | 0.02 | 0.02 | 0.005 | 0.02 | None |
| Variable | var131 | 0.03 | 0.03 | 0.005 | 0.03 | None |
| Variable | var150 | 0.02 | 0.02 | 0.005 | 0.02 | None |
| Dependent | var200 | Definition: (var100+var130)/2 | | | | |
| Constant | var500 | 1.2 | | | | |

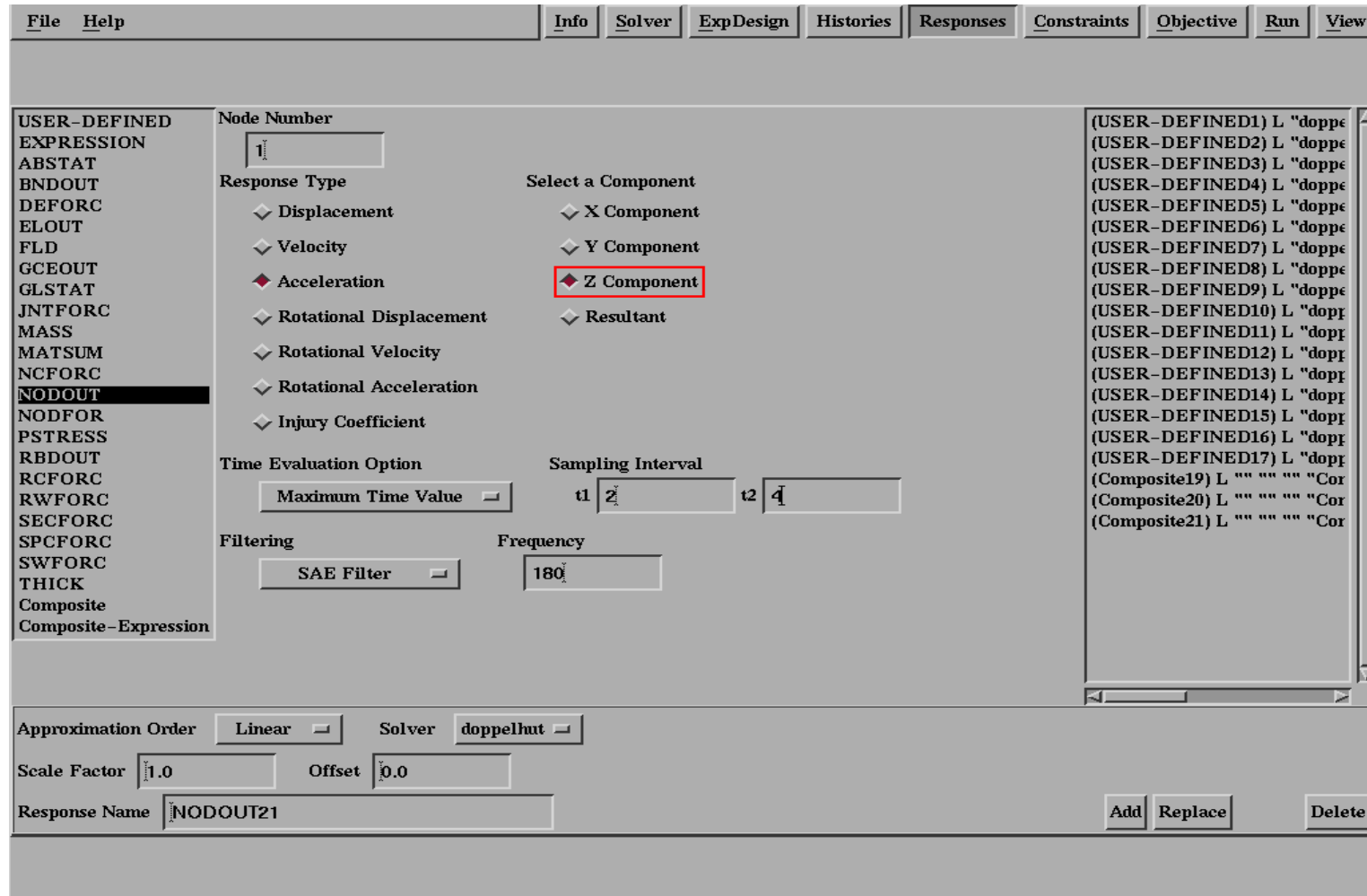
Add a Variable Delete Variable

Optimization using the Successive Response Surface Method



LS-OPT: Application of the SRSM

Graphical User Interface

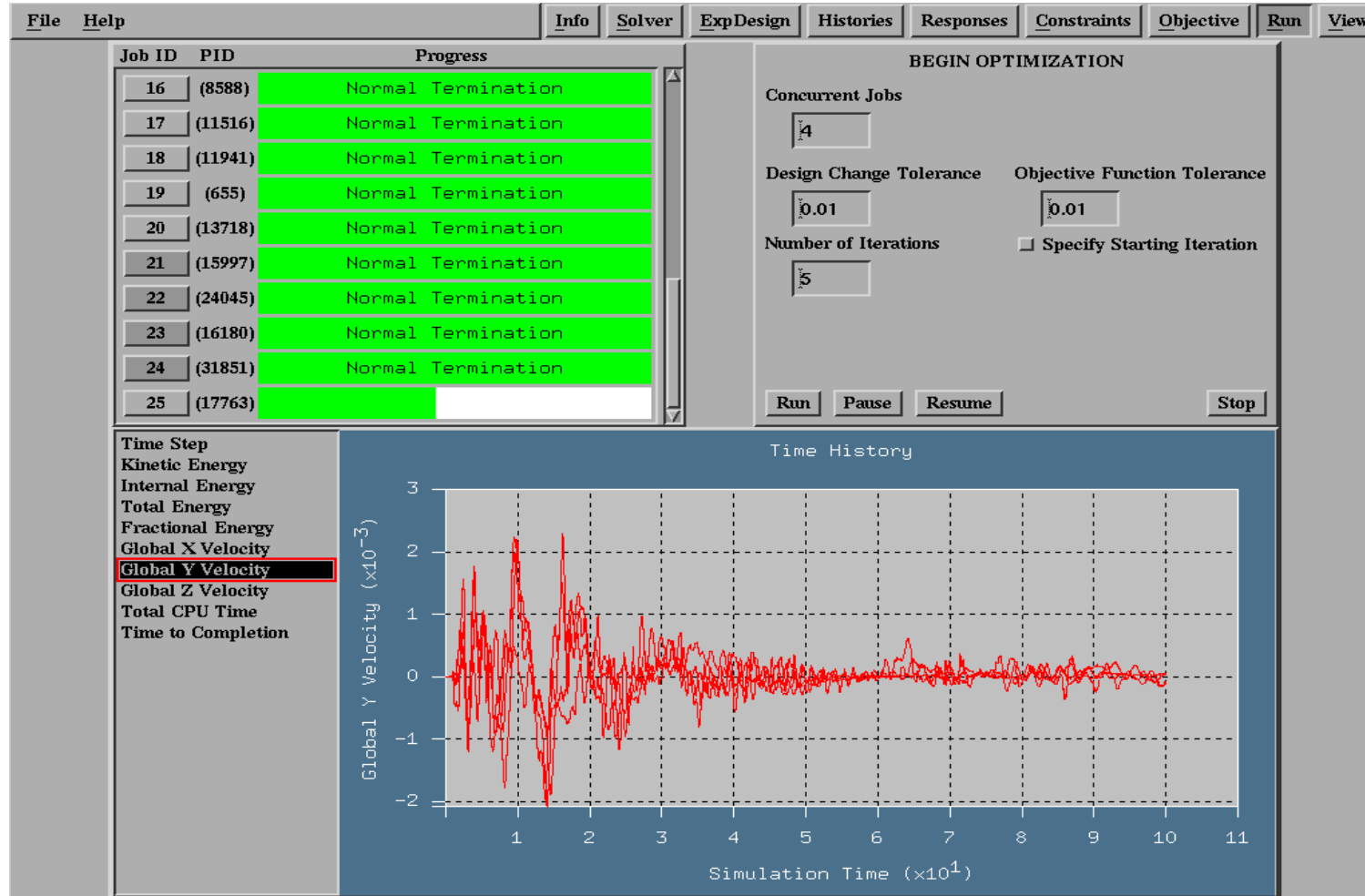


Optimization using the Successive Response Surface Method



LS-OPT: Application of the SRSM

Graphical User Interface



Optimization using the Successive Response Surface Method



LS-OPT: Application of the SRSM

Advantages of the Method

➤ **Global Optimization:**

Response Surface have a tendency to capture globally optimal regions. Local minima caused by noisy response as well as the step-size dilemma for numerical gradients are avoided

➤ **Parallel Computation:**

Successive Response Surface scheme allows parallel (independent) computation of experimental points within one iteration

➤ **Flexible Design Exploration:**

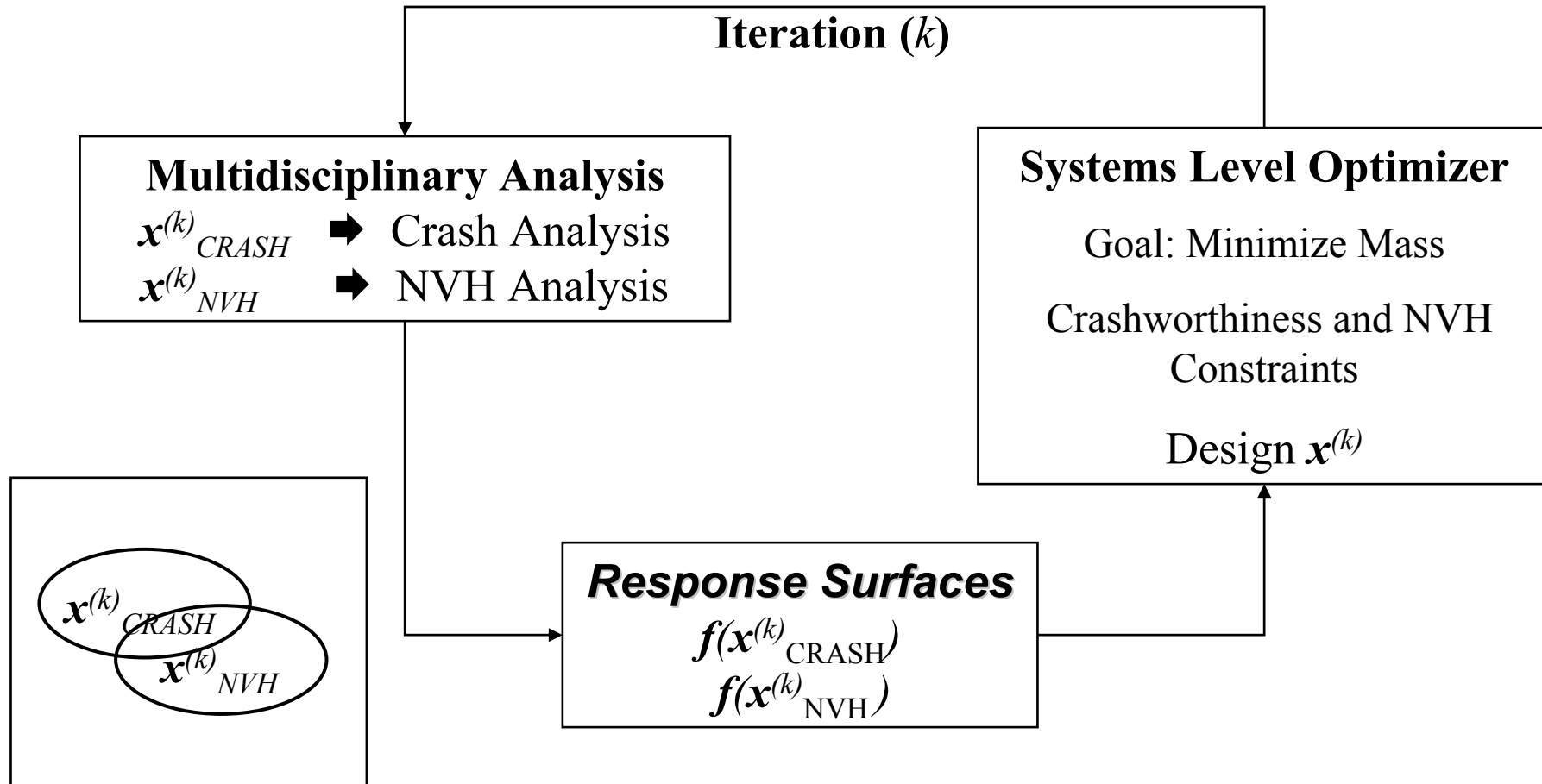
Design exploration can be changed within the optimization process. Thus, control of the computational time and the quality of the Response Surface is possible

➤ **Trade-Off Studies:**

Since the Response Surface is determined, easy examination of varying constraint bounds is possible (not reliable with linear approximations)

Example: Multidisciplinary Optimization (MDO)

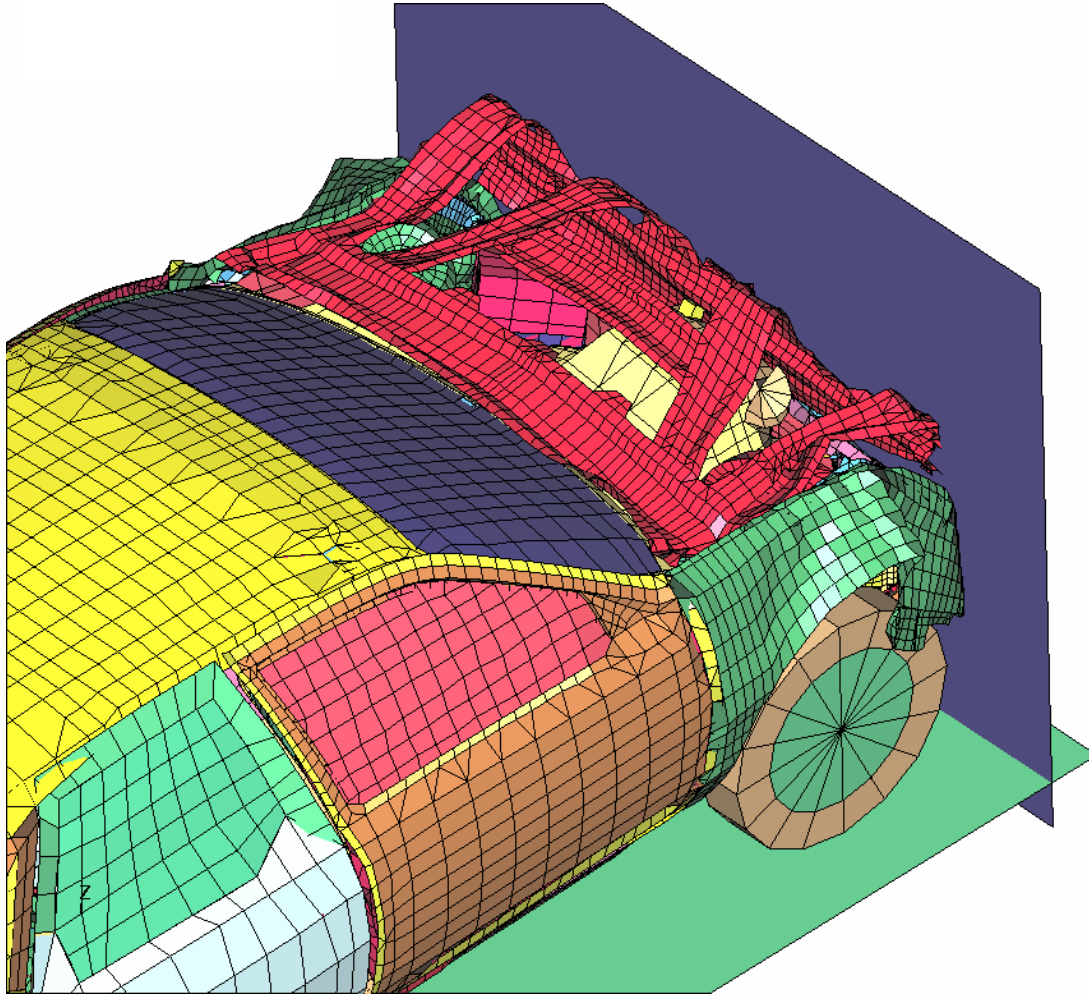
Fully Integrated Optimization - Crash and NVH



Optimization using the Successive Response Surface Method

Example: Multidisciplinary Optimization (MDO)

Full Vehicle – Crash Performance (LS-DYNA)



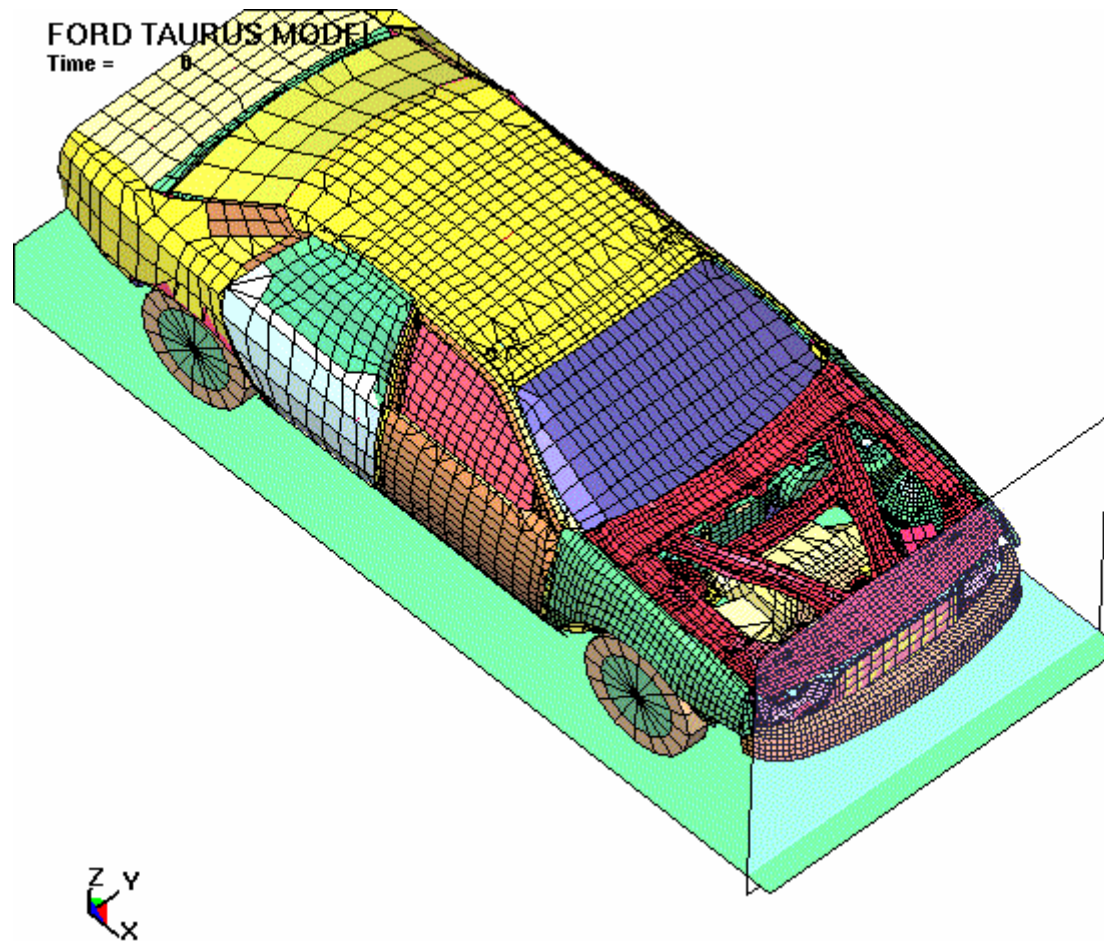
Baseline:

- **30 000 elements**
- **Displacement = 552mm**
- **Stage1Pulse = 14.34 g**
- **Stage2Pulse = 17.57 g**
- **Stage3Pulse = 20.76 g**

Optimization using the Successive Response Surface Method

Example: Multidisciplinary Optimization (MDO)

Full Vehicle – Crash Performance (LS-DYNA)



Optimization using the Successive Response Surface Method

DYNA
MORE

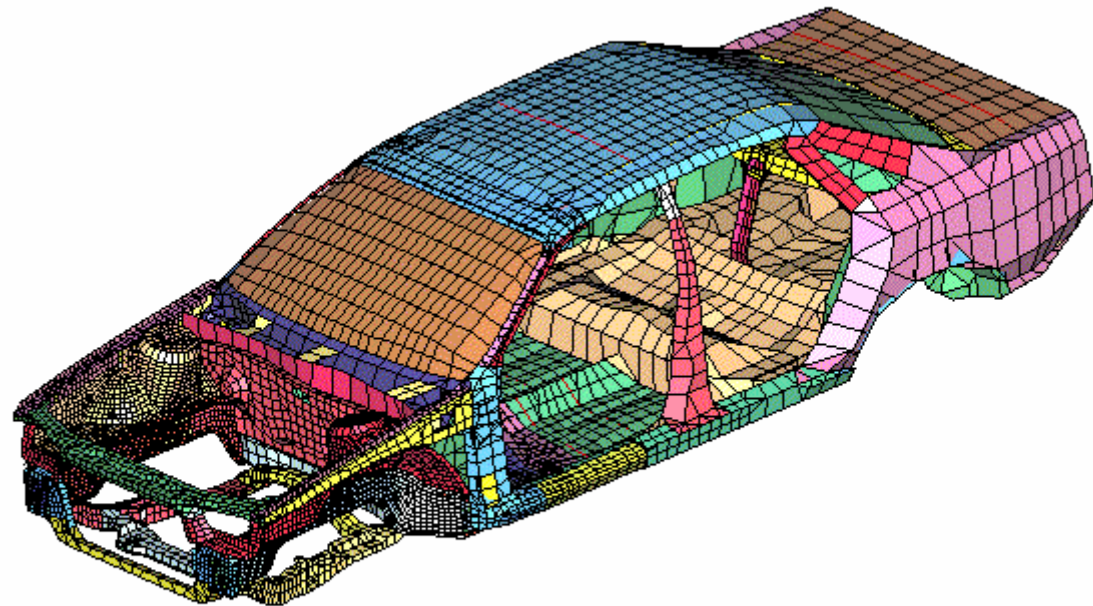
Example: Multidisciplinary Optimization (MDO)

BIW-Modell - NVH Performance (LS-DYNA)

LS-DYNA eigenvalue problem - FORD TAURUS BIW
Time = 38.736

Baseline:

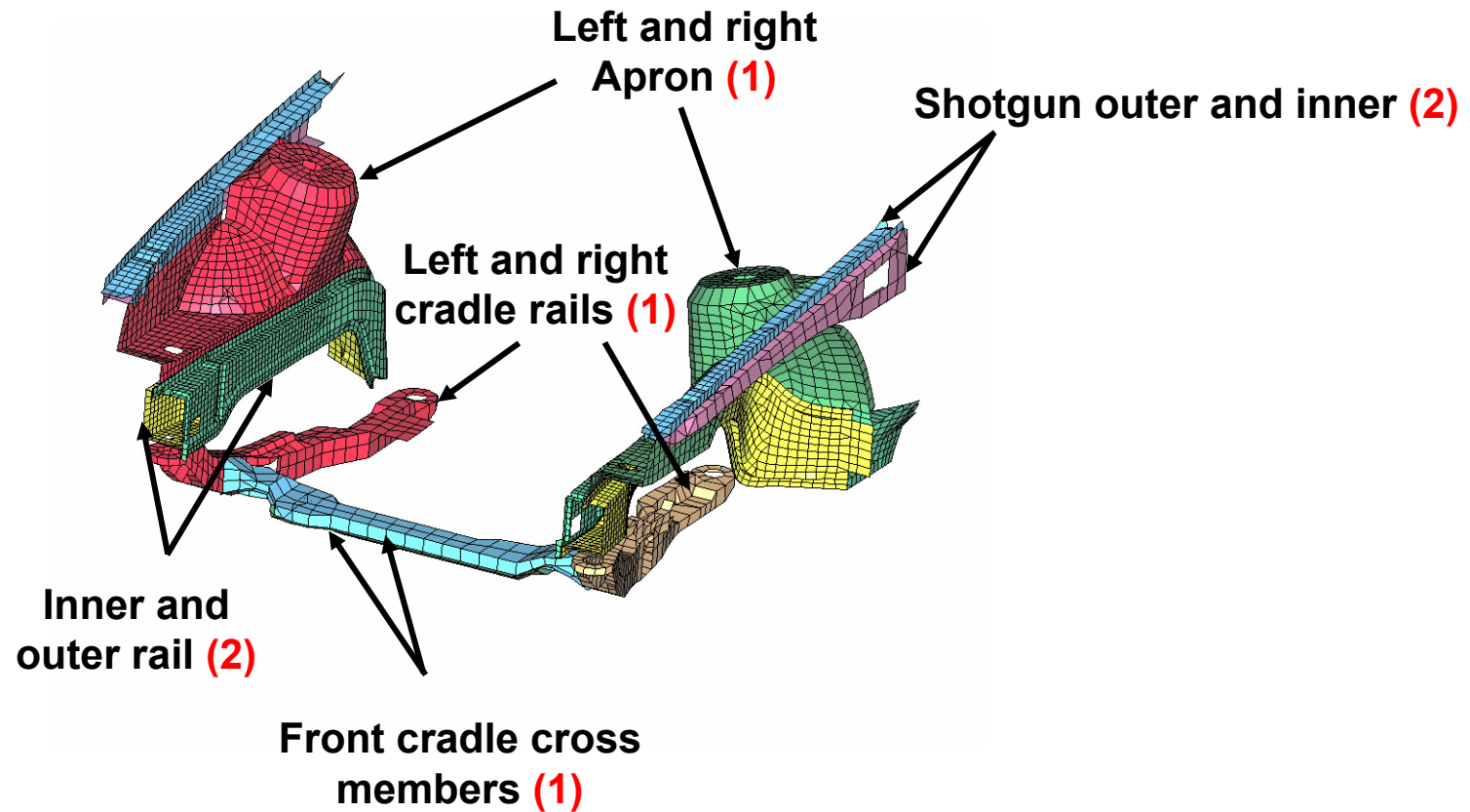
- **18 000 elements**
- **Torsional Mode 1**
Frequency = 38.7 Hz



Optimization using the Successive Response Surface Method

Example: Multidisciplinary Optimization (MDO)

Design Variables (Thickness)



Optimization using the Successive Response Surface Method

Example: Multidisciplinary Optimization (MDO)

Design Formulation – FULLY SHARED VARIABLES

Design Objective:

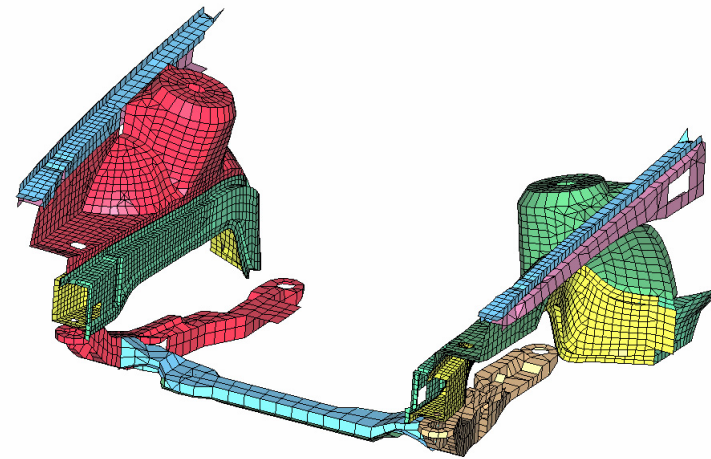
- Minimize (Mass of components)

Design Constraints:

- Displacement > 551.8mm
- $37.77\text{Hz} < \text{Torsional mode 1 frequency} < 39.77\text{Hz}$
- Stage1Pulse > 14.34g
- Stage2Pulse > 17.57g
- Stage3Pulse > 20.76g

Thickness Design Variables Shared: 7

- Rails (inner and outer),
Shotgun (inner and outer), Aprons,
Cradle rails, cross member



Example: Multidisciplinary Optimization (MDO)

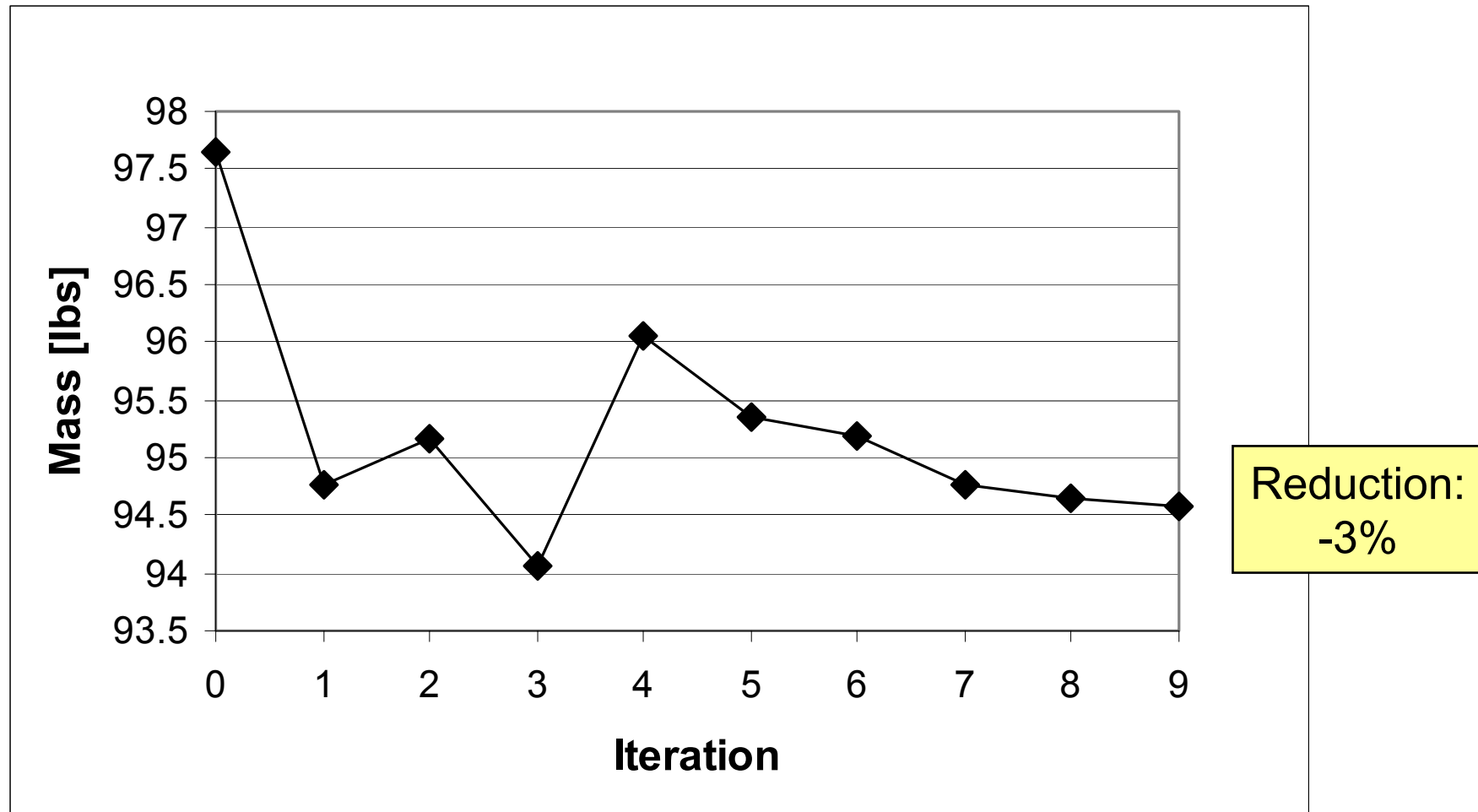
Mode Tracking

- During NVH optimization necessary to **track mode** as mode switching can occur due to design changes
- **Search for maximum scalar (dot) product** between eigenvector of base mode and each solved mode:

$$\max_j \left\{ \left(M_0^{\frac{1}{2}} \phi_0 \right)^T \left(M_j^{\frac{1}{2}} \phi_j \right) \right\}$$

Example: Multidisciplinary Optimization (MDO)

Optimization History: Mass (Objective) – FULLY SHARED VARIABLES

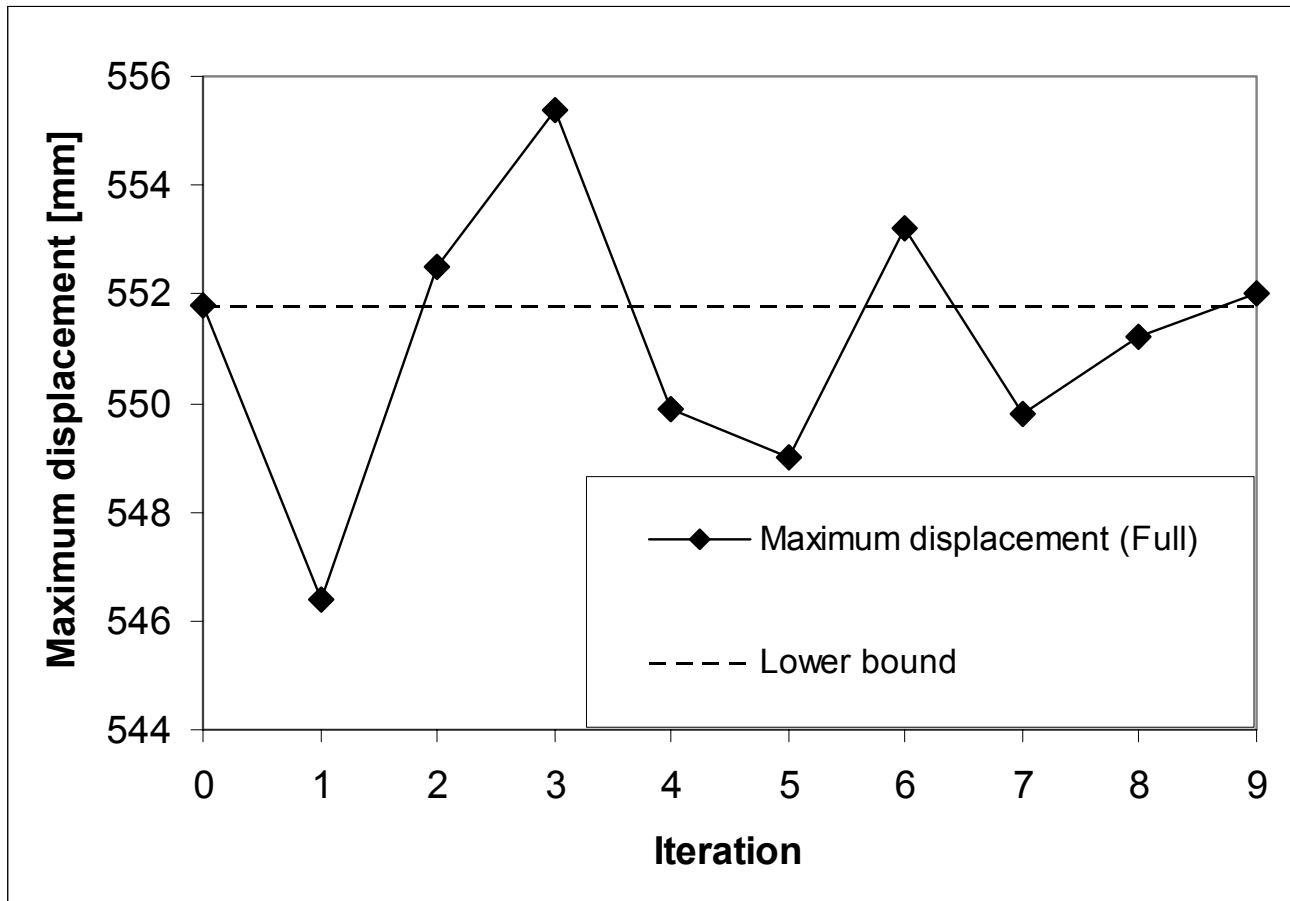


Optimization using the Successive Response Surface Method



Example: Multidisciplinary Optimization (MDO)

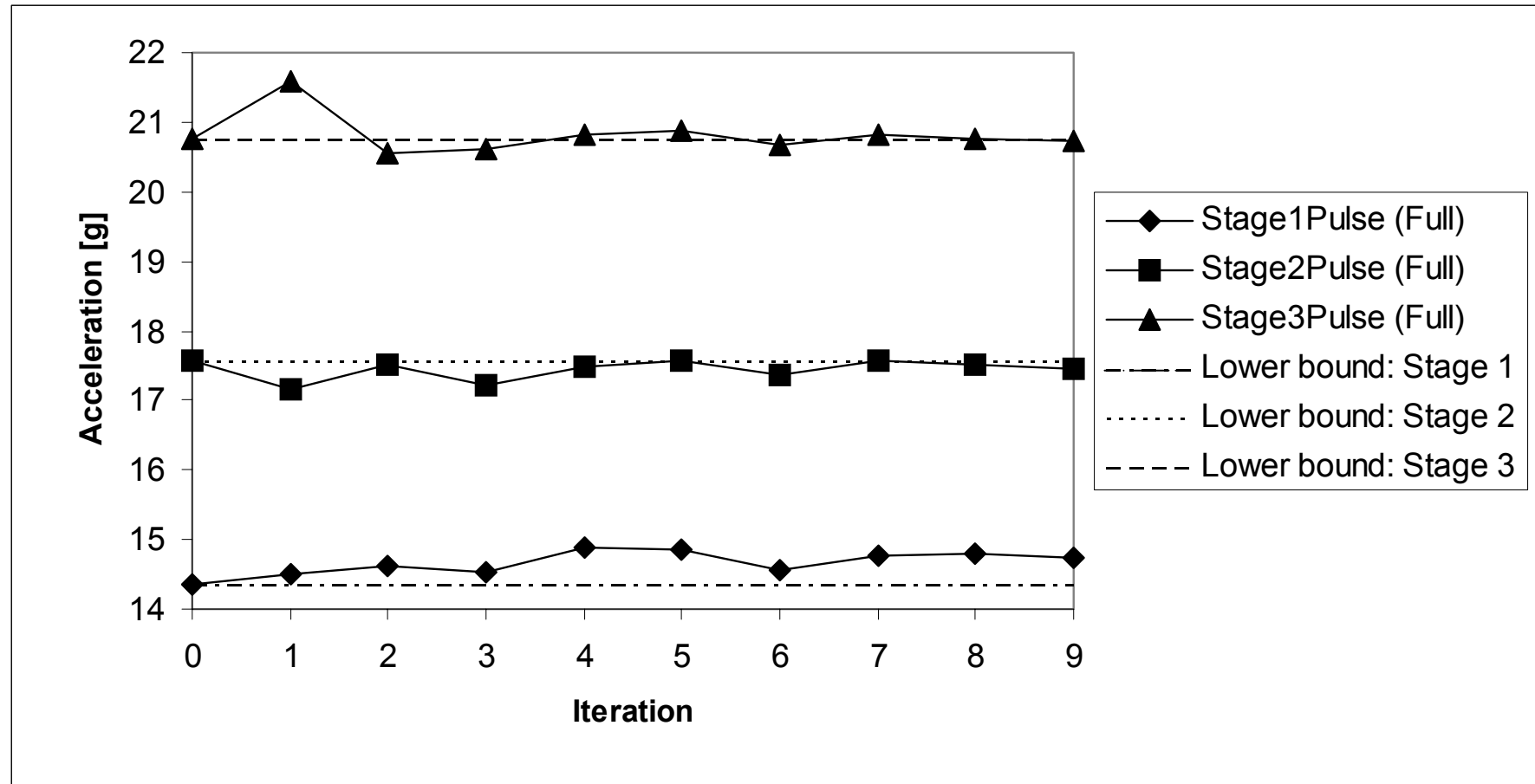
Optimization History: Maximum Displacement – FULLY SHARED VARIABLES



Optimization using the Successive Response Surface Method

Example: Multidisciplinary Optimization (MDO)

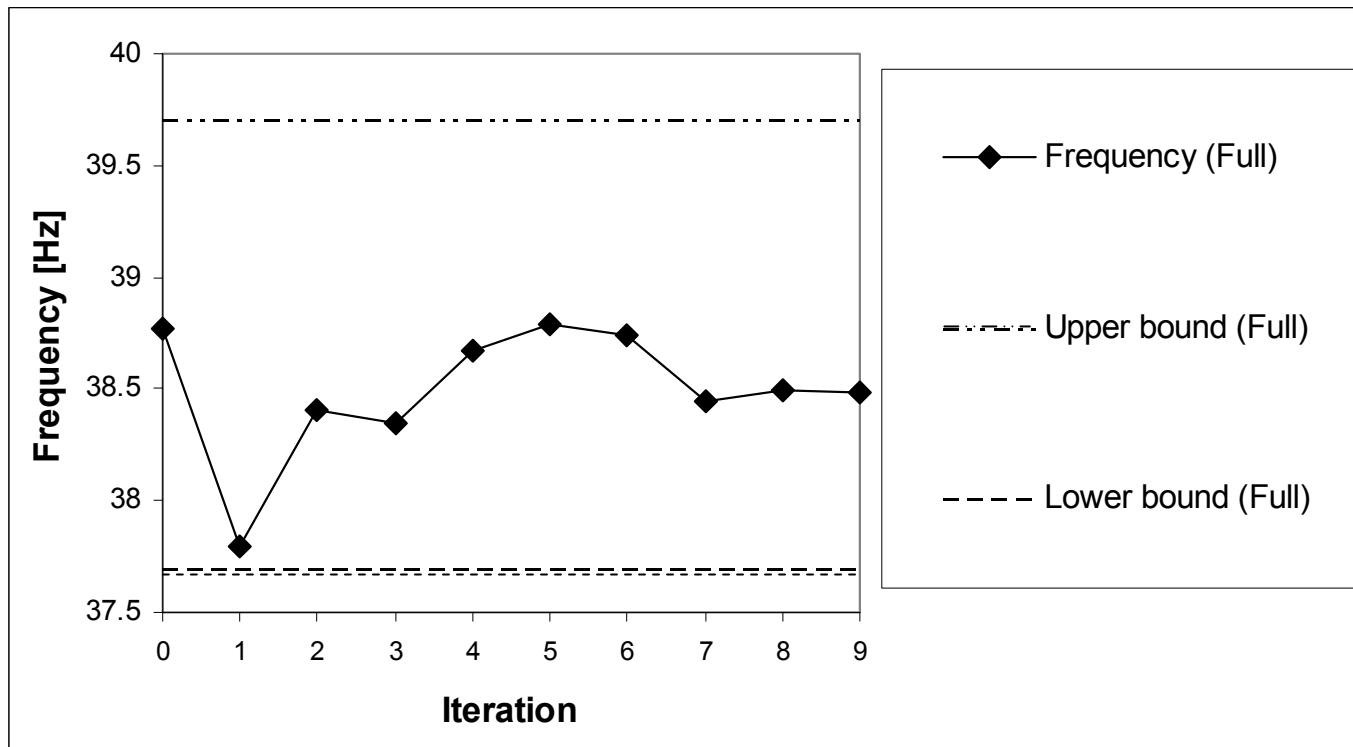
Optimization History: Stage Pulses – FULLY SHARED VARIABLES



Optimization using the Successive Response Surface Method

Example: Multidisciplinary Optimization (MDO)

Optimization History: Torsional Mode Frequency – FULLY SHARED VARIABLES

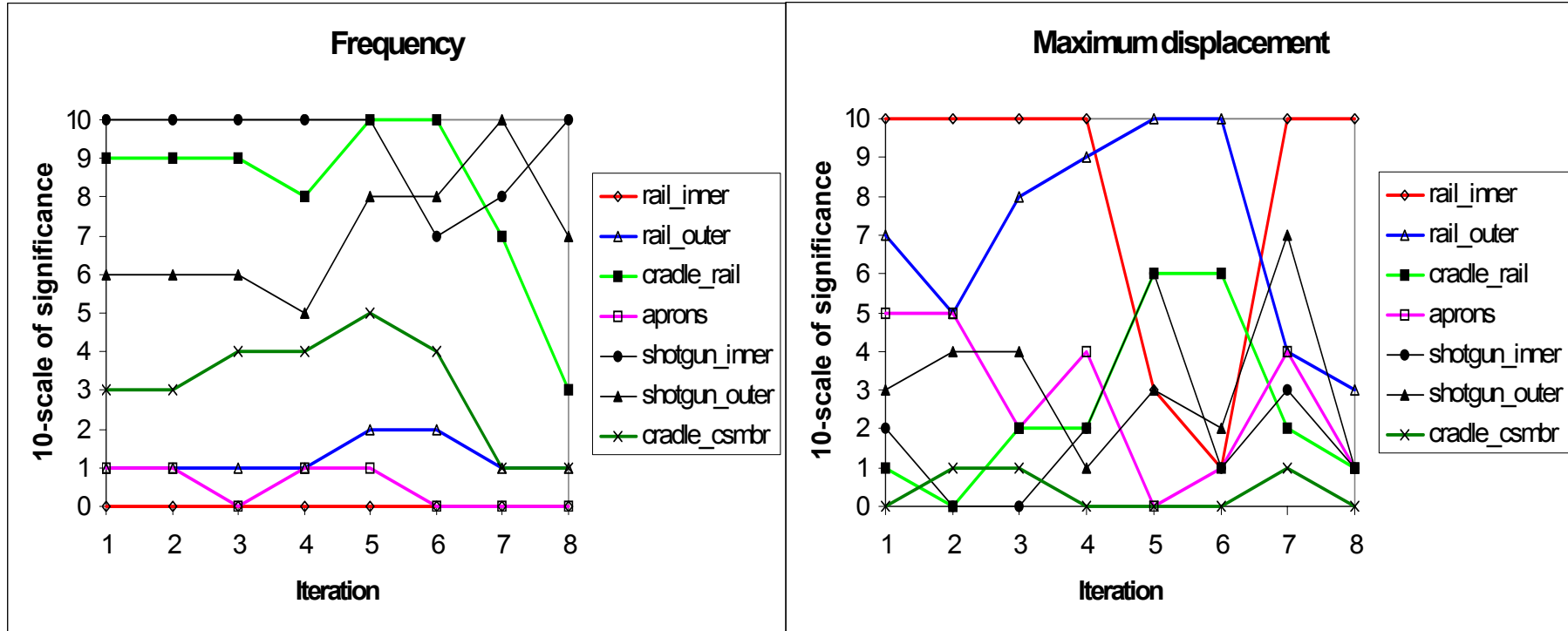


Optimization using the Successive Response Surface Method



Example: Multidisciplinary Optimization (MDO)

Variable Screening

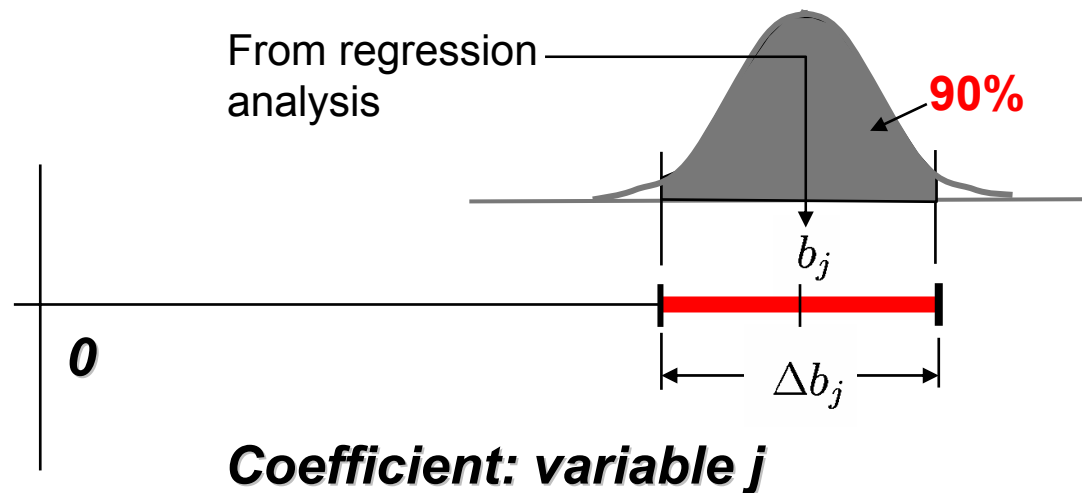


➤ **Goal: Remove of less significant variables**

Example: Multidisciplinary Optimization (MDO)

Variable Screening

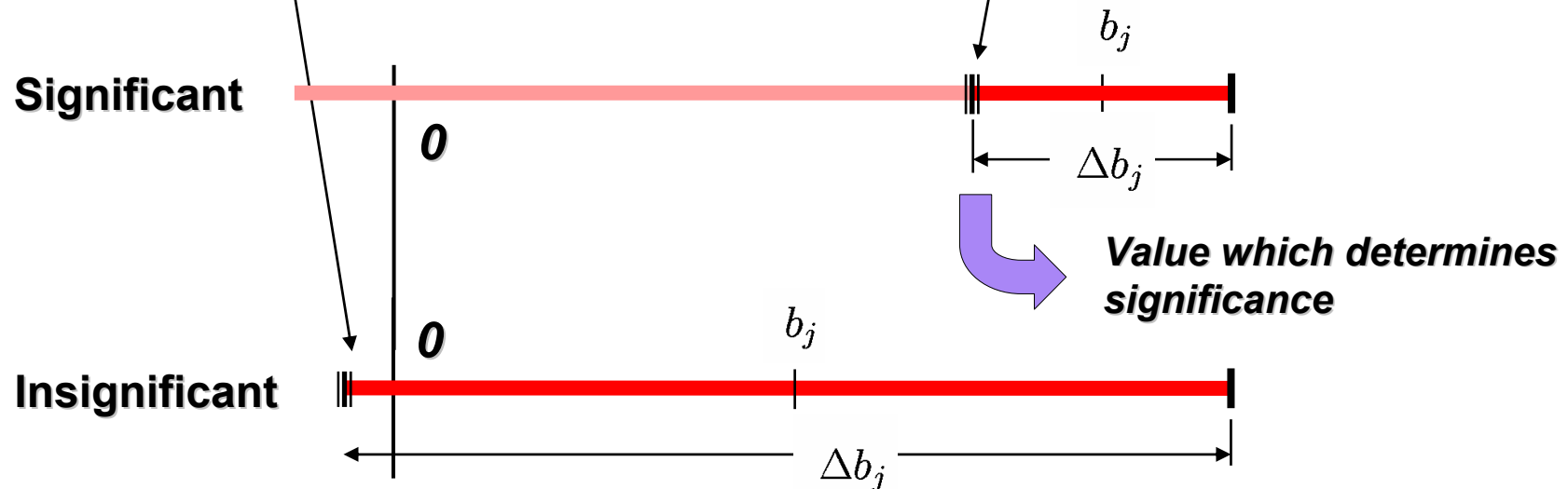
- **Methodology: ANOVA (ANalysis Of VAriance)**
- Δb_j depends on **the variance** of the simulation points
- **Use a 90% confidence level** and determine the lower bound



Example: Multidisciplinary Optimization (MDO)

Variable Screening

- Variables are **ranked** according to **lower bound**
- If the **lower bound < 0** , regression coefficient is **insignificant**
- In a linear approximation, a **variable can be removed** if its coefficient is insignificant



Example: Multidisciplinary Optimization (MDO)

Design Formulation – PARTIALLY SHARED VARIABLES

Design Objective:

- Minimize (Mass of Components)

Design Constraints:

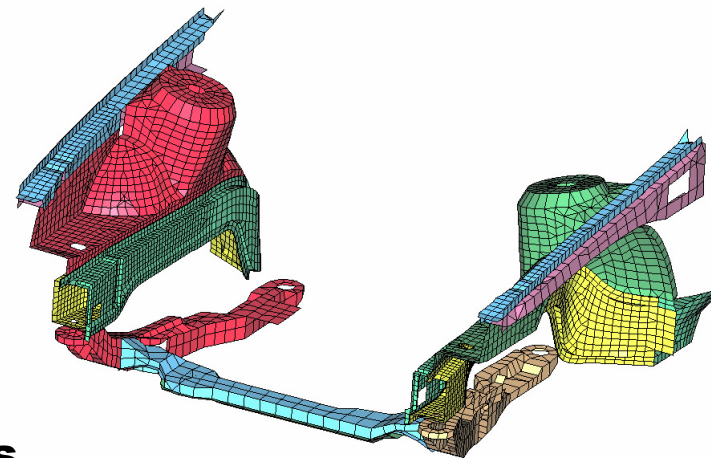
- Displacement > 551.8mm
- $38.27\text{Hz} < \text{Torsional Mode 1 frequency} < 39.27\text{Hz}$
- Stage1Pulse > 14.34g
- Stage2Pulse > 17.57g
- Stage3Pulse > 20.76g

Crashworthiness Design Variables: 6

- Rails (inner and outer), Shotgun (inner and outer), Aprons, Cradle Rails

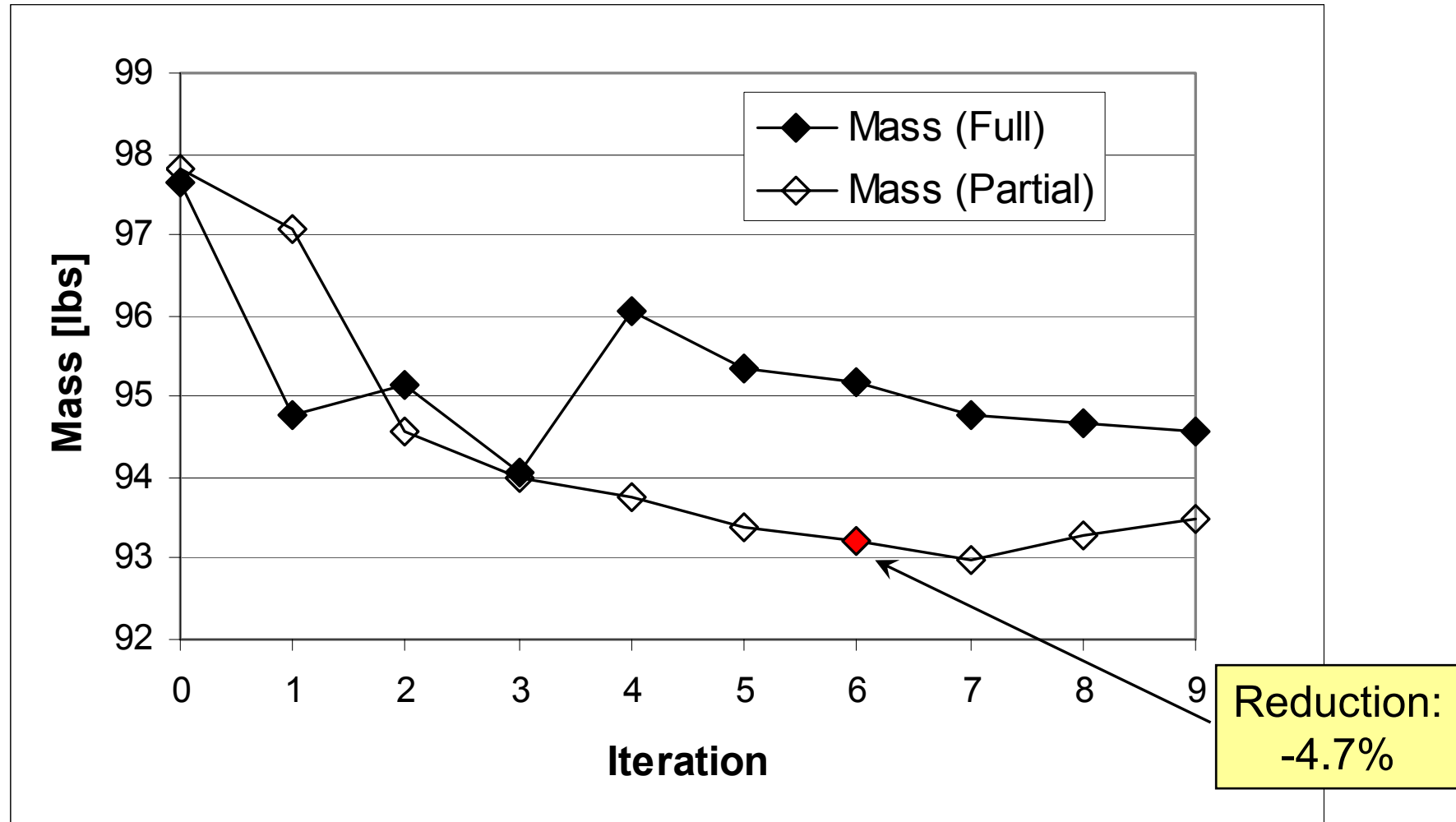
NVH Design Variables: 4

- Shotgun (inner and outer), Cradle Rails, Cross Member



Example: Multidisciplinary Optimization (MDO)

Optimization History: Mass (Objective)

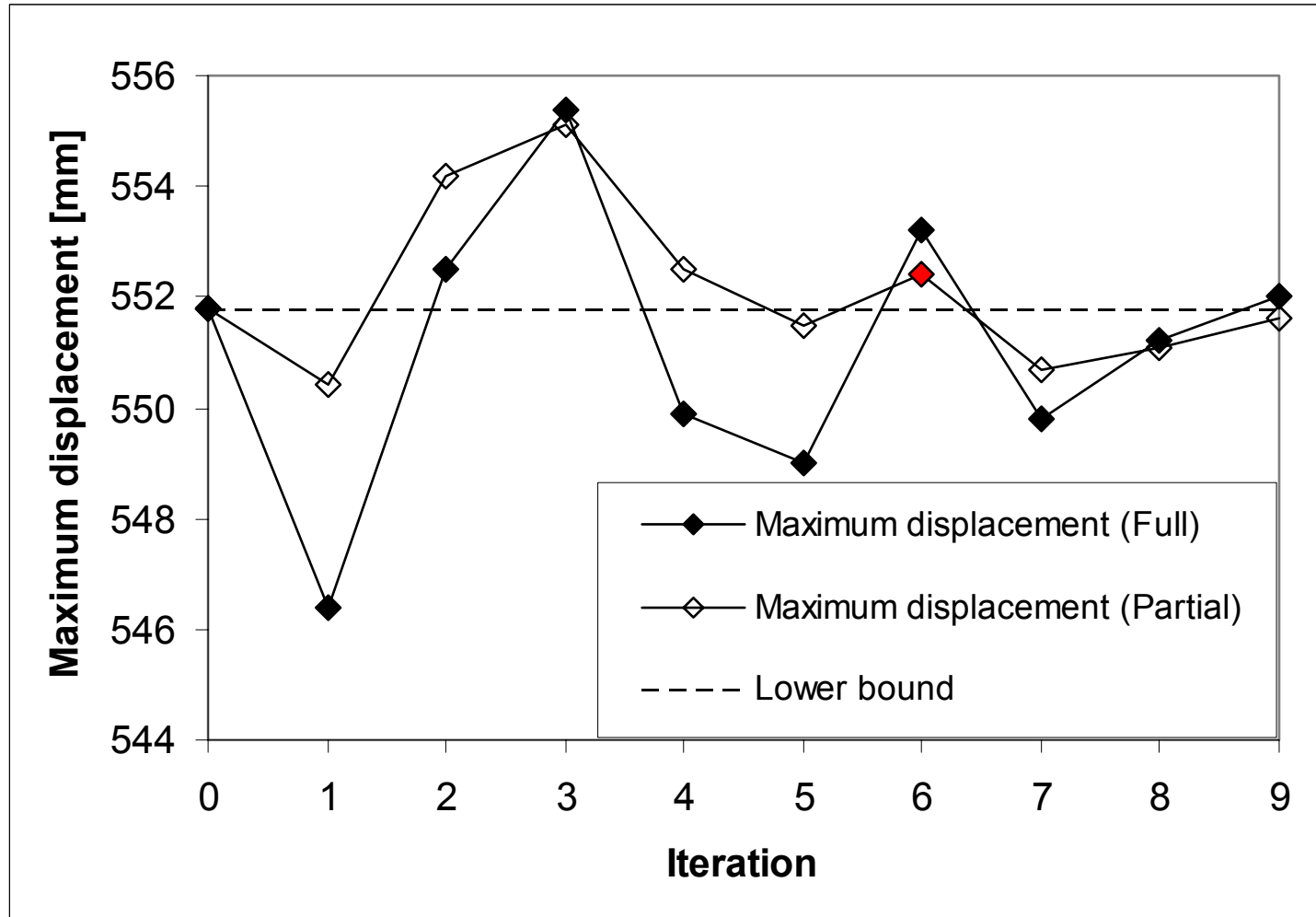


Optimization using the Successive Response Surface Method



Example: Multidisciplinary Optimization (MDO)

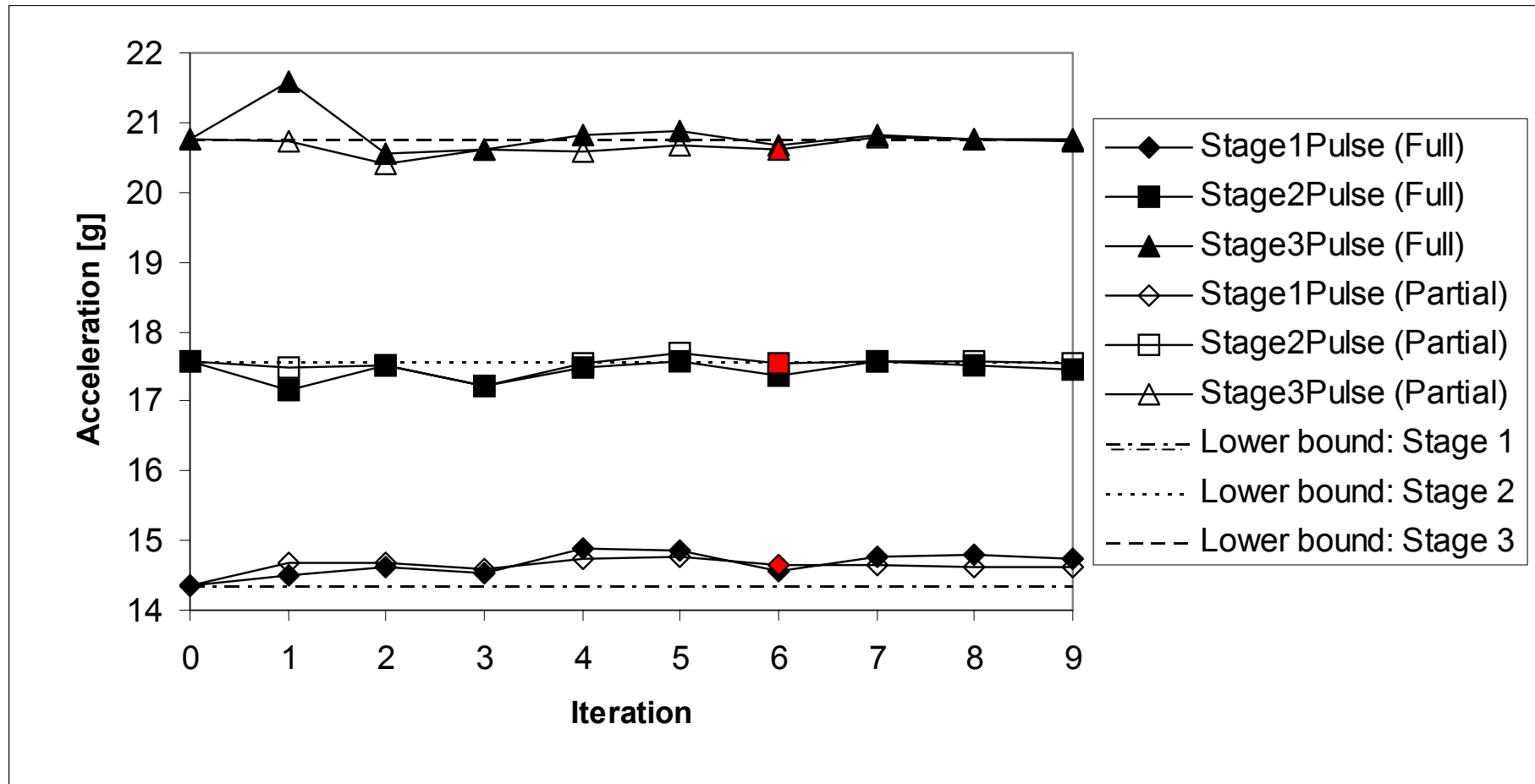
Optimization History: Maximum Displacement



Optimization using the Successive Response Surface Method

Example: Multidisciplinary Optimization (MDO)

Optimization History: Stage Pulses

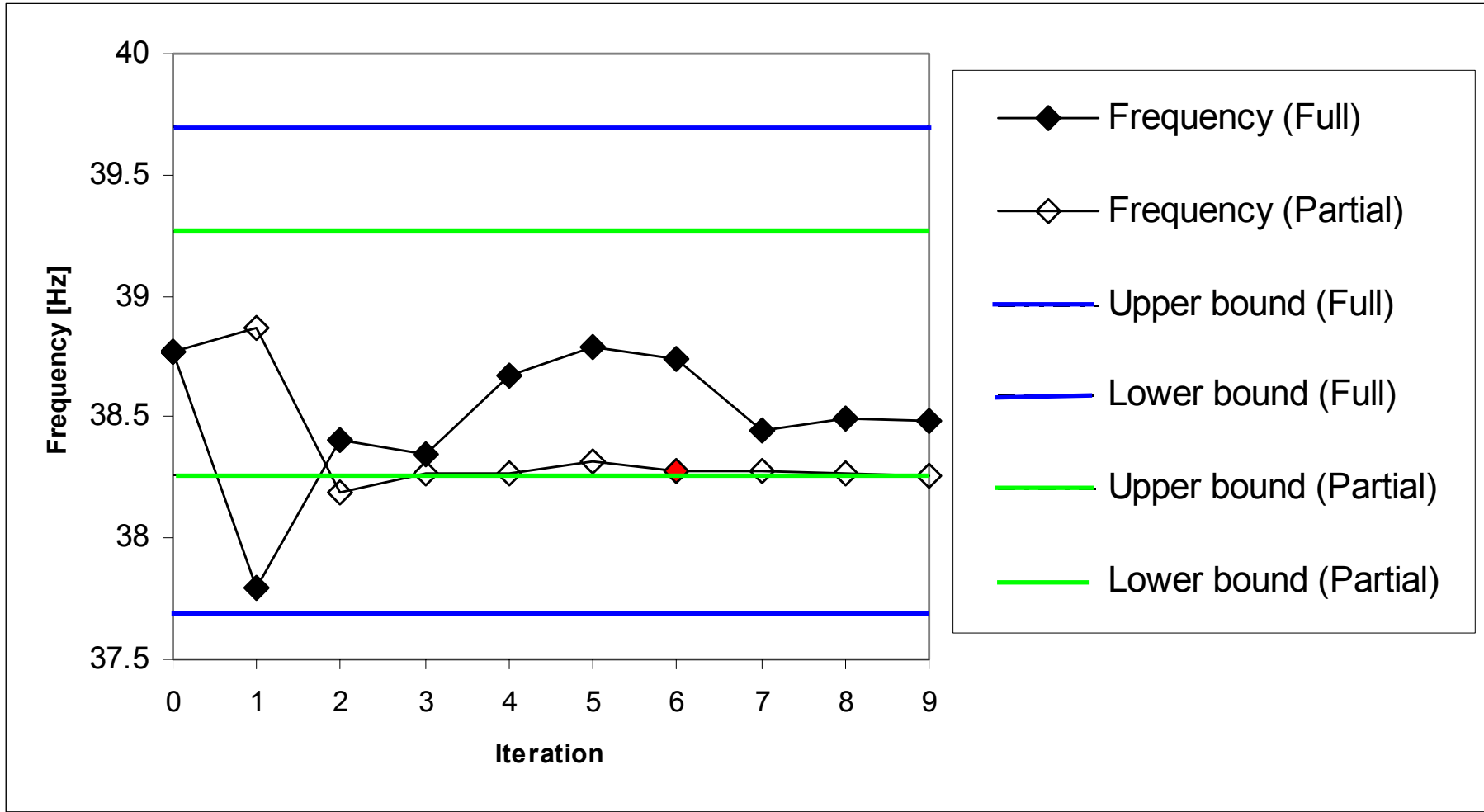


Optimization using the Successive Response Surface Method



Example: Multidisciplinary Optimization (MDO)

Optimization History: Torsional Frequency



Optimization using the Successive Response Surface Method



Example: Multidisciplinary Optimization (MDO)

Run Statistics

Run Statistics – Fully Shared MDO

13 experimental points per iteration per discipline

- **7 hours per crash simulation**
- **10 minutes per NVH simulation (700MB memory each)**
- **9 iterations to converge**
- **117 crash simulations and 117 NVH simulations**

Run Statistics – Partially Shared MDO

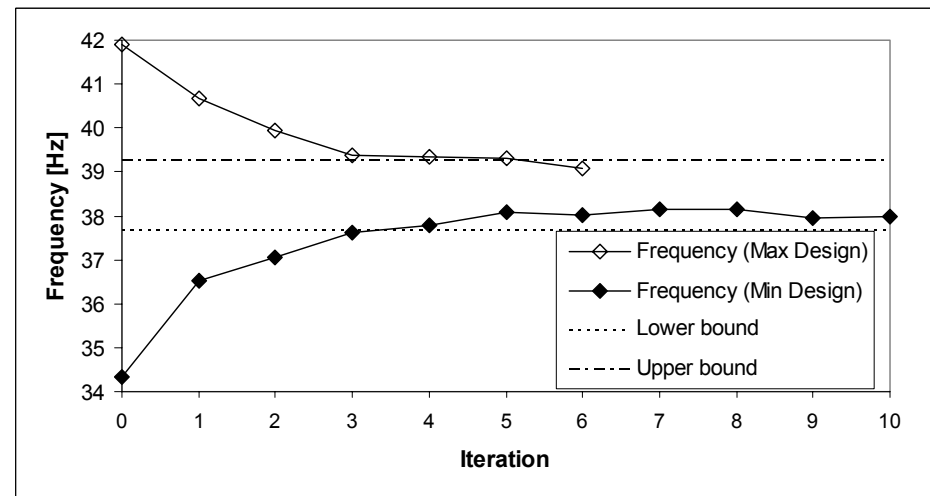
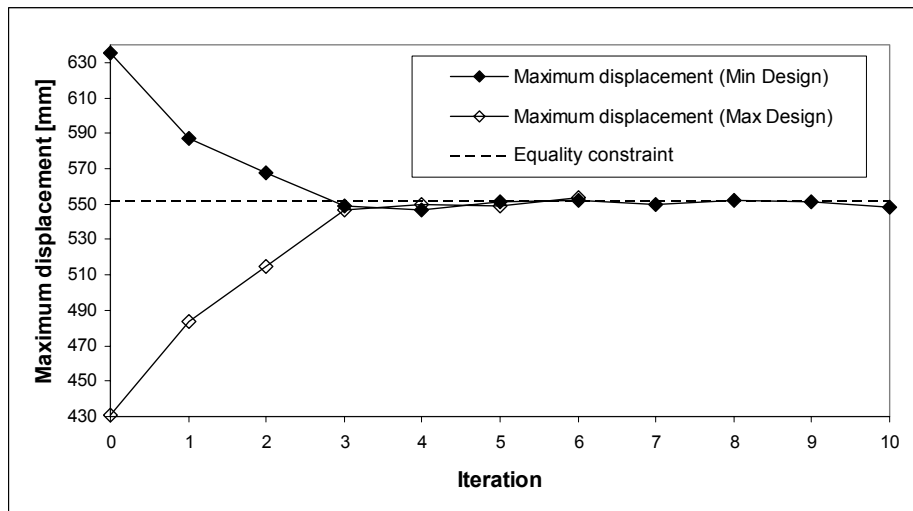
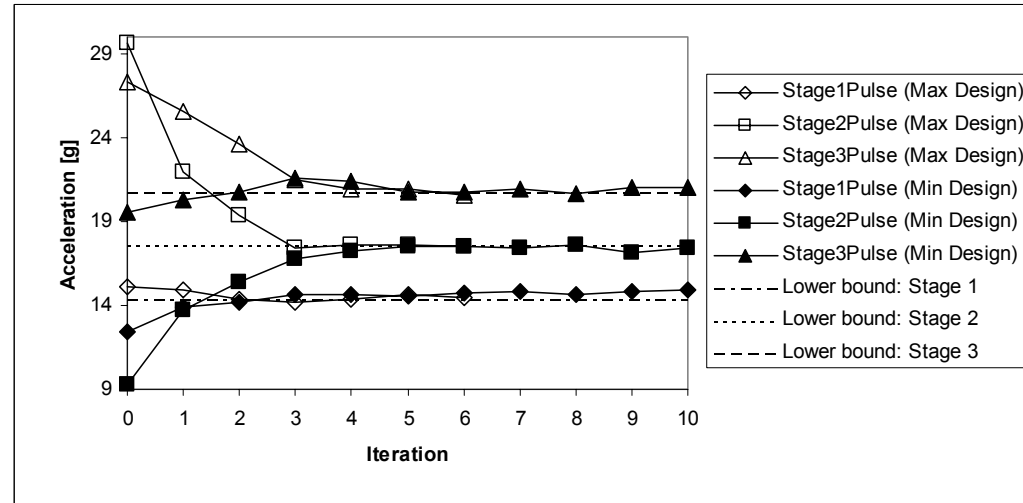
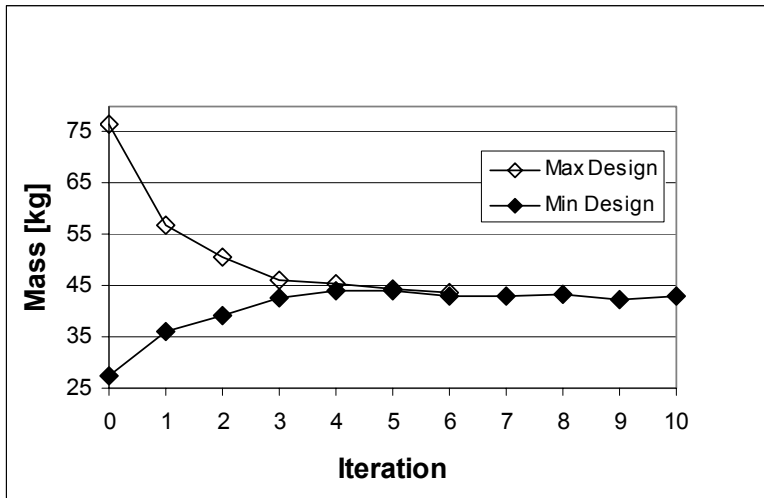
11 experimental points per iteration for crash

8 experimental points per iteration for NVH

- **6 iterations for good compromised solution**
- **66 crash simulations and 48 NVH simulations**
- **More flexibility in using resources (processors and memory)**

Example: Multidisciplinary Optimization (MDO)

Starting from Lightest and Heaviest Design



Optimization using the Successive Response Surface Method



Conclusions MDO-Example

Conclusions / Outlook / Remarks

- **Multidisciplinary feasible optimization** of a full vehicle model considering crashworthiness and NVH design criteria is described
- Almost **5% mass reduction** is achieved while maintaining or improving of the design criteria of the baseline design
- **Variable Screening** allows the detection of insignificant design variables
- The capability of **partially or non shared variables for MDO** may reduce the computational effort dramatically

Conclusions MDO-Example

Conclusions / Outlook / Remarks

- **Optimization with current full vehicle crash models (500000-1000000 Elements) is still very time consuming and **requires huge hardware resources****
- **Gradients of the linear implicit discipline (NVH) may be used for the calculation of the according Response Surface approximation**
- **Discrete Methodologies for sheet thickness optimization**
- **A two-stage approach with **stochastic and deterministic methods**, may be very efficient for crash**