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Isogeometric Shell Components in Full Vehicle Crash Simulations: Hybrid Modeling

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Frank Bauer BMW Group, Munich, Germany BMW GROUP





- 1. Introduction to Hybrid Modeling
- 2. Isogeometric Shell Components in Full Vehicle Crash Simulations
 - 2.1 Handling of Trimmed Multi-Patch NURBS Shells
 - 2.2 Analysis Capabilities for Explicit Dynamic Crash
 - 2.3 Connection Modeling
 - 2.4 Process-specific Capabilities
- 3. Examples: Hybrid Vehicle Crash Simulations
 - 3.1 Front Crash
 - 3.2 Side Crash
- 4. Conclusion and Outlook





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Why Isogeometric Analysis?

Potential benefits

- Faster development process by integrating design and analysis
 - Same NURBS-based geometry description, consistent data structure
 - Mesh-independent modeling (spotwelds, connections, etc.)
- Higher predictive accuracy (for similar element size)
 - More accurate geometry description: Consider details neglected with standard FEA
 - Higher-order AND higher-continuity basis
 - Smooth solution field
 - Capture deformation modes correctly

Increased efficiency

- Larger element size and fewer DOFs (for similar accuracy)
- Larger time step size in explicit dynamics (for similar element size, C^{p-1} continuity and interior elements)





Courtesy of BMW Group



Linear FEA





Why Hybrid IGA/FEA Models?

- Long-established and optimized development processes for FEA
- All-encompassing IGA process: Requires fundamental changes and mind shift
- 1. Impossible to change entire process at once
- 2. Pure isogeometric vehicle model not yet possible
- \rightarrow Start with small changes: Replace certain components
- \rightarrow Make changing to IGA as simple as changing the element formulation
- \rightarrow Build trust in the technology
- \rightarrow Demonstrate specific benefits











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Handling of Trimmed Multi-Patch NURBS Shells

LS-DYNA Capabilities

- 1. Processing of CAD data: Geometry + Topology + Analysis information
 - → CAD-inspired ***IGA** keyword family

*IGA Keywords for Geo	ometry and Topology
*IGA_1D_NURBS_UVW	NID NR PR RKI U V W WGT
*IGA_EDGE_UVW	EID EXYZID NID
*IGA_EDGE_XYZ	EID NID PSID
*IGA_1D_BREP	BRID EIDi
*IGA_2D_NURBS_XYZ	NID NR NS PR PS RKI SKI X Y Z WGT
*IGA_FACE_XYZ	FID NID ORI PSID ESID BRIDi
*IGA_SHELL	SID PID NISR NISS
*PART	PID SECID MID
*SECTION_IGA_SHELL	SECID ELFORM SHRF NIP IRL
*MAT_ELASTIC	MID RO E PR



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Handling of Trimmed Multi-Patch NURBS Shells

LS-DYNA Capabilities

- 1. Processing of CAD data: Geometry + Topology + Analysis information
- 2. Numerical integration of trimmed NURBS elements [1]
- 3. Application of coupling and boundary conditions [2,3,4]
- 4. Stabilization of small trimmed elements [4]



[1] A.P. Nagy & D.J. Benson, On the numerical integration of trimmed isogeometric elements. Comput. Methods Appl. Mech. Eng. 284 (2015), 165–185.

[2] M. Breitenberger, A. Apostolatos, B. Philipp, R. Wüchner, K.-U. Bletzinger, Analysis in computer aided design: Nonlinear isogeometric B-Rep analysis of shell structures, Comput. Methods Appl. Mech. Eng. 284 (2015) 401–457.

[3] L.F. Leidinger, M. Breitenberger, A.M. Bauer, S. Hartmann, R. Wüchner, K.-U. Bletzinger, F. Duddeck, L. Song, Explicit dynamic isogeometric B-Rep analysis of penalty-coupled trimmed NURBS shells, Comput. Methods Appl. Mech. Eng. 351 (2019) 891–927.
 [4] L.F. Leidinger, Explicit Isogeometric B-Rep Analysis for Nonlinear Dynamic Crash Simulations: Integrating Design and Analysis by Means of Trimmed Multi-Patch Shell Structures, PhD thesis, Technical University of Munich, Germany (2020).



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Time Step Size, Time Step Estimation, Mass Scaling



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Time Step Size, Time Step Estimation, Mass Scaling

Bar models Effect of trimmed element size on time step size? **Basis functions** \rightarrow Practically, no effect for IGA with C^{p-1} and p > 1Linear FEA Boundary vs. interior elements? NURBS p=2 $\rightarrow \Delta t_{\text{Interior}} > \Delta t_{\text{Boundary}} \rightarrow \text{Cut-off boundaries}$ "Extend" option NURBS p=3 in ANSA Mode Create NURBS p=4 Distortion di... 0.05 Subinterval par. ✓ Min span Max span 20. ¹¹trim ω_{\max}^e Extend Join $\Delta t_{\rm crit}$ light control points $\omega_{\rm max}$ inactive control points [4]

[4] L. F. Leidinger, Explicit Isogeometric B-Rep Analysis for Nonlinear Dynamic Crash Simulations: Integrating Design and Analysis by Means of Trimmed Multi-Patch Shell Structures, PhD thesis, Technical University of Munich, Germany (2020).

Time Step Size, Time Step Estimation, Mass Scaling

- Effect of trimmed element size on time step size?
 → Practically, no effect for IGA with C^{p-1} and p > 1
- Boundary vs. interior elements?

 $\rightarrow \Delta t_{\text{Interior}} > \Delta t_{\text{Boundary}} \rightarrow \text{Cut-off boundaries}$

- Accurate time step estimation
 - \rightarrow IGADO=1 in ***CONTROL_TIMESTEP**
 - \rightarrow Account for continuity







Time Step Size, Time Step Estimation, Mass Scaling

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- Accurate time step estimation
 - → IGADO=1 in *CONTROL_TIMESTEP
 → Account for continuity
- Mass scaling for predefined time step
 - \rightarrow Significant scaling for FEA with same mesh size

IGA time step estimate **not accounting** for continuity: → Significant mass scaling



IGA time step **accounting** for continuity: → No mass scaling



Plasticity, Damage and Failure

- Goal: No modification of existing material data
- Plasticity: Use existing elasto-plastic material models with $E, \nu, \sigma(\varepsilon^p, \dot{\varepsilon})$
- Damage and Failure modeling (e.g. DIEM)
 - Continuum Damage Mechanics approach $\sigma = (1 D) \tilde{\sigma}$
 - 1. Damage initiation D > 0 determined by Forming Limit Curve
 - 2. Damage evolution $\dot{D} = \dot{D}(\dot{\varepsilon}^p, \eta, D, \mathbf{l})$
 - 3. Integration point failure if D = 1 char. element length
 - 4. Element failure: If n in-plane IPs and m layers failed

Control points

*MAT ADD DAMAGE DIEM

DIEM = Damage Initiation and Evolution Model

Damage variable





Plasticity, Damage and Failure

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 - 2. Damage evolution $\dot{D} = \dot{D}(\dot{\varepsilon}^p, \eta, D, l)$
 - 3. Integration point failure if D = 1 char. element length
 - 4. Element failure: If n in-plane IPs and m layers failed
 - Crack modeling
 - So far: simple element deletion
 - \rightarrow Discontinuity? \rightarrow "Cross-talk"!
 - \rightarrow Delete *p* elements for discontinuity
 - More sophisticated crack modeling to be developed





*MAT_ADD_DAMAGE_DIEM DIEM = Damage Initiation and Evolution Model

Control points

Integration points

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Damage variable



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Connection Modeling

For One-To-One Component Exchange (FEA→IGA)

- Spotwelds: PID-based and mesh-independent
 - *CONSTRAINED_INTERPOLATION_SPOTWELD





Connection Modeling

For One-To-One Component Exchange (FEA→IGA)

- Spotwelds: PID-based and mesh-independent
 - *CONSTRAINED_INTERPOLATION_SPOTWELD
- Tied contact between bolts and IGA shells (PID-based)
 - *CONTACT_TIED_SHELL_EDGE_TO_SURFACE_BEAM_OFFSET



*CON	STRAINED	INTERPOLAT	TION_SPOTW	ELD		
\$	MID	SID	NSID	THICK	R	
	101	102	200	3.0	7	
\$	RN	RS	BETA1	LCF	LCUPF	
	XXX	XXX	XXX	XXX	XXX	
\$	ES	EB	ET	LCDEXP	GAMMA	
	XXX	XXX	XXX	XXX	XXX	



Connection Modeling

For One-To-One Component Exchange (FEA→IGA)

- Spotwelds: PID-based and mesh-independent
 - *CONSTRAINED_INTERPOLATION_SPOTWELD
- Tied contact between bolts and IGA shells (PID-based)
 - *CONTACT_TIED_SHELL_EDGE_TO_SURFACE_BEAM_OFFSET
- Attach existing rigid bodies (pins and bolts)
 - "Glue" FE nodes of ***CNRB** to IGA shells using
 - *IGA_POINT_UVW

*IGA	_POINT_UVW	I		
\$	IDP	IDN	υI	VI
	1	3001	0.78	0.15



STRAINED	INTERPOLA	TION_SPOTW	ELD	
MID	SID	NSID	THICK	R
101	102	200	3.0	7
RN	RS	BETA1	LCF	LCUPF
XXX	XXX	XXX	XXX	XXX
ES	EB	ET	LCDEXP	GAMMA
XXX	XXX	XXX	XXX	XXX
	STRAINED MID 101 RN XXX ES XXX	STRAINED_INTERPOLAT MID SID 101 102 RN RS XXX XXX ES EB XXX XXX	STRAINED_INTERPOLATION_SPOTWMIDSID101102200RNRSBETA1XXXXXXESEBXXXXXX	STRAINED_INTERPOLATION_SPOTWELDMIDSIDNSIDTHICK1011022003.0RNRSBETA1LCFXXXXXXXXXXXXESEBETLCDEXPXXXXXXXXXXXX





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Process-Specific Capabilities

For One-To-One Component Exchange (FEA→IGA)

- Initialization with material history data (e.g. from forming simulations)
 - Current approach
 - External mapping from dynain file using DYNAmore ENVYO
 - Difficulty: IGA integration point location for trimmed elements
 - *INITIAL_STRESS/STRAIN_IGA_SHELL

Possible future approach

- Internal mapping in LS-DYNA
- Spatial point cloud + field data for history variables



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Overview

- Conventional FEA vehicle model
 - \rightarrow Replace certain components with IGA shells
- One-to-one component exchange without further modifications
- Semi-automatic IGA model generation with ANSA
- IGA settings:
 - Cubic NURBS-based Reissner-Mindlin shell
 - 4mm average element length
 - Reduced Gauss integration (3x3 integration points)
 - New time step estimate → practically no mass scaling for IGA shells
 - Plasticity, damage and failure considered





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Front Crash: Car Against Rigid Wall

- Hybrid model with IGA longitudinal members
 - Two trimmed single-patch shells each side
 - Connected via spotwelds + bolts
 - ~5.5M elements
- Full width frontal impact





50 %

Front Crash: Car Against Rigid Wall

Pure FEA vs. Hybrid IGA/FEA model

Side view











Front Crash: Car Against Rigid Wall

Pure FEA vs. Hybrid IGA/FEA model

Top view (overlay)

Right load path





Left load path





Courtesy or Divivy Group



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Side Crash: Car Against Rigid Pole

- Hybrid model with IGA seat cross-members
 - Basis + top component
 - Spotweld and bolt connections
 - Including dummy + airbags (~9M elements)

Basis component single trimmed patch

32 km/h 75° 254 mm Pole







Side Crash: Car Against Rigid Pole

Pure FEA vs. Hybrid IGA/FEA model

Oblique view





Hybrid IGA/FEA







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Conclusion

Hybrid Modeling

- Industrial workflow ANSA LS-DYNA enables
 - Hybrid IGA/FEA full vehicle crash simulations
 - Simple 1:1 exchange of shell components
 - \rightarrow Ready for productive tests
- Next Steps
 - Increase numerical efficiency + robustness
 - Validation of damage and failure, connection technology
- Ongoing R&D topics
 - Improved modeling of cracks and discontinuities
 - Feature-based modeling
 - Trimmed IGA solids





1



Thank You

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